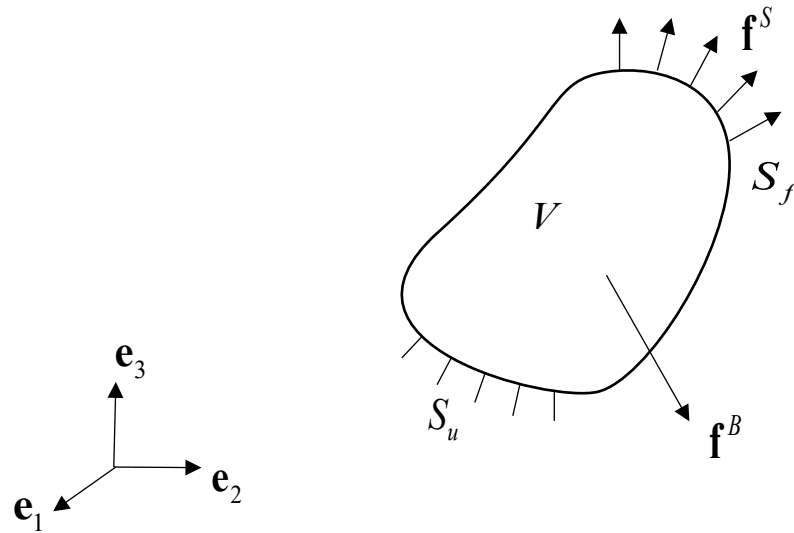


## 4. Finite Element Formulation



$$\text{PVW: } \int_V \tau_{ij} \delta \varepsilon_{ij} dV = \int_V f_i^B \delta u_i dV + \int_{S_f} f_i^S \delta u_i dS$$

Let us assume

- ① Small displacement:  $V_o \approx V$
- ② Linear elastic material:  $\tau_{ij} = C_{ijkl} \varepsilon_{kl}$

In the finite element formulation,

$$\boldsymbol{\tau} = \begin{bmatrix} \tau_{11} \\ \tau_{22} \\ \tau_{33} \\ \tau_{12} \\ \tau_{23} \\ \tau_{31} \end{bmatrix}, \quad \boldsymbol{\varepsilon} = \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ \gamma_{12} \\ \gamma_{23} \\ \gamma_{31} \end{bmatrix},$$

where  $\gamma_{ij} = 2\varepsilon_{ij}$  ( $i \neq j$ ) : engineering shear strain

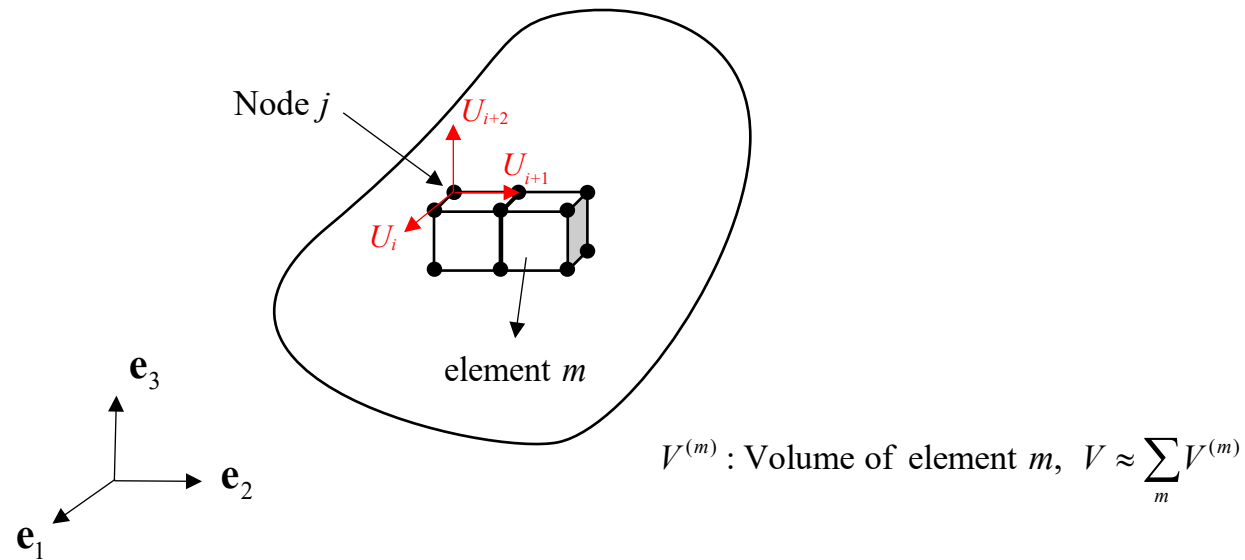
$$\tau_{ij}\varepsilon_{ij} = \sum_{i=1}^3 \sum_{j=1}^3 \tau_{ij}\varepsilon_{ij} = \boldsymbol{\varepsilon}^T \boldsymbol{\tau} = \boldsymbol{\varepsilon} \cdot \boldsymbol{\tau} \quad (\text{note: } \tau_{12}\varepsilon_{12} + \tau_{21}\varepsilon_{21} = 2\tau_{12}\varepsilon_{12} = \tau_{12}\gamma_{12})$$

Material Law:  $\boldsymbol{\tau} = \mathbf{C}\boldsymbol{\varepsilon}$

PVW in vector/matrix form:

$$\boxed{\int_V \delta \boldsymbol{\varepsilon}^T \mathbf{C} \boldsymbol{\varepsilon} dV = \int_V \delta \mathbf{u}^T \mathbf{f}^B dV + \int_{S_f} \delta \mathbf{u}^T \mathbf{f}^S dS}$$

## Finite Element Discretization



Let us assume that  $S_f = S$  and  $S_u = \phi \rightarrow$  "PVW still works."

Nodal displacement at node  $j \rightarrow \begin{bmatrix} U_i \\ U_{i+1} \\ U_{i+2} \end{bmatrix}$

Nodal displacement vector (nodal DOFs vector),

$$\mathbf{U} = \begin{bmatrix} U_1 \\ \vdots \\ U_i \\ U_{i+1} \\ \vdots \\ U_N \end{bmatrix}, \quad N: \text{number of the total DOFs}$$

$$\text{PVW: } \sum_m \int_{V^{(m)}} \delta \boldsymbol{\varepsilon}^{(m)\top} \mathbf{C}^{(m)} \boldsymbol{\varepsilon}^{(m)} dV^{(m)} = \sum_m \int_{V^{(m)}} \delta \mathbf{u}^{(m)\top} \mathbf{f}^{B(m)} dV^{(m)} + \sum_m \int_{S_1^{(m)} \dots S_q^{(m)}} \delta \mathbf{u}^{s(m)\top} \mathbf{f}^{S(m)} dS^{(m)},$$

where  $S_1^{(m)} \dots S_q^{(m)}$  are surfaces of "element  $m$ " on boundary.

## Displacement Interpolations

$$\mathbf{u}^{(m)} = \mathbf{H}^{(m)}\mathbf{U} \quad (\text{interpolation of displacement})$$

$$\delta\mathbf{u}^{(m)} = \mathbf{H}^{(m)}\delta\mathbf{U} \quad (\text{interpolation of virtual displacement})$$

where

$\mathbf{u}^{(m)}$  : displacement field of element  $m$

$\delta\mathbf{u}^{(m)}$  : virtual displacement field of element  $m$

$\mathbf{U}$  : nodal displacement vector

$\delta\mathbf{U}$  : virtual nodal displacement vector

$\mathbf{H}^{(m)}$  : displacement interpolation matrix for element  $m$

(Note) The same interpolation is used for real and virtual displacements.

→ "symmetric stiffness matrix"

From the strain-displacement relation,

$$\varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \quad \rightarrow \quad \boldsymbol{\varepsilon}^{(m)} = \mathbf{B}^{(m)} \mathbf{U}$$

$$\delta \varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial \delta u_i}{\partial x_j} + \frac{\partial \delta u_j}{\partial x_i} \right) \quad \rightarrow \quad \delta \boldsymbol{\varepsilon}^{(m)} = \mathbf{B}^{(m)} \delta \mathbf{U}$$

$\boldsymbol{\varepsilon}^{(m)}$  : strain field of element m

$\delta \boldsymbol{\varepsilon}^{(m)}$  : virtual strain field of element m

$\mathbf{B}^{(m)}$  : strain interpolation matrix for element m

Using the displacement and strain interpolations in PVW, the following equation is obtained

$$\delta \mathbf{U}^T \left[ \sum_m \int_{V^{(m)}} \mathbf{B}^{(m)T} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)} \right] \mathbf{U} = \delta \mathbf{U}^T \left[ \sum_m \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{B(m)} dV^{(m)} + \sum_m \int_{S_1^{(m)} \dots S_q^{(m)}} \mathbf{H}_s^{(m)T} \mathbf{f}^{S(m)} dS^{(m)} \right]$$

(eq. 3.1)

Substituting  $\delta\mathbf{U} = [1 \ 0 \ \dots \ 0]^T$  ( $\delta U_1 = 1$ , others = 0) into (eq. 3.1), we obtain a linear equation

$$K_{11}U_1 + K_{12}U_2 + \dots + K_{1N}U_N = R_1$$

Substituting  $\delta\mathbf{U} = [0 \ 1 \ 0 \ \dots \ 0]^T$  ( $\delta U_2 = 1$ , others = 0) into (eq. 3.1), we obtain another equation

$$K_{21}U_1 + K_{22}U_2 + \dots + K_{2N}U_N = R_2$$

We can do this task  $N$  times.

This process to apply "virtual displacement vectors" is the same to

$$\cancel{\mathcal{I}} \left[ \sum_m \int_{V^{(m)}} \mathbf{B}^{(m)T} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)} \right] \mathbf{U} = \cancel{\mathcal{I}} \left[ \sum_m \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{B(m)} dV^{(m)} + \sum_m \int_{S_1^{(m)} \dots S_q^{(m)}} \mathbf{H}_s^{(m)T} \mathbf{f}^{S(m)} dS^{(m)} \right]$$

Finally, we obtain a set of  $N$  linear equations.  $\rightarrow$  "static equilibrium equations"

$$\boxed{\mathbf{K}\mathbf{U} = \mathbf{R}}$$

where  $\mathbf{K} = \sum_m \mathbf{K}^{(m)}$  with  $\mathbf{K}^{(m)} = \int_{V^{(m)}} \mathbf{B}^{(m)\top} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)}$

$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S \quad \left\{ \begin{array}{l} \mathbf{R}_B = \sum_m \mathbf{R}_B^{(m)} \quad \text{with } \mathbf{R}_B^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)\top} \mathbf{f}^{B(m)} dV^{(m)} \\ \mathbf{R}_S = \sum_m \mathbf{R}_S^{(m)} \quad \text{with } \mathbf{R}_S^{(m)} = \int_{S_1^{(m)} \dots S_q^{(m)}} \mathbf{H}_s^{(m)\top} \mathbf{f}^{S(m)} dS^{(m)} \end{array} \right.$$

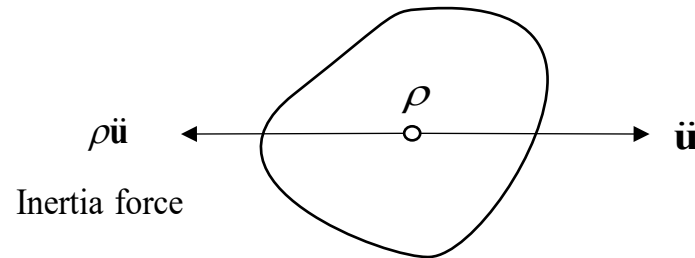
**K**: Stiffness matrix

**U**: Nodal displacement vector

**R**: Nodal force vector



## Dynamic Equilibrium Equations (Equations of motion)



$$\mathbf{R}^{B(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)\top} \mathbf{f}^{B(m)} dV^{(m)}$$

$$\mathbf{f}^{B(m)} = \bar{\mathbf{f}}^{B(m)} + (-\rho \ddot{\mathbf{u}}^{(m)}) \quad \leftarrow \quad \mathbf{u}^{(m)} = \mathbf{H}^{(m)} \mathbf{U} \quad \text{and} \quad \ddot{\mathbf{u}}^{(m)} = \mathbf{H}^{(m)} \ddot{\mathbf{U}}$$

$$\mathbf{R}^{B(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)\top} \bar{\mathbf{f}}^{B(m)} dV^{(m)} - \left[ \int_{V^{(m)}} \mathbf{H}^{(m)\top} \rho^{(m)} \mathbf{H}^{(m)} dV^{(m)} \right] \ddot{\mathbf{U}}$$

$$\mathbf{K}\mathbf{U} = \mathbf{R} - \mathbf{M}\ddot{\mathbf{U}} \quad \text{with "mass matrix" } \mathbf{M} = \sum_m \mathbf{M}^{(m)}, \quad \mathbf{M}^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)\top} \rho^{(m)} \mathbf{H}^{(m)} dV^{(m)}$$

$$\mathbf{M}\ddot{\mathbf{U}} + \mathbf{K}\mathbf{U} = \mathbf{R}$$

## Imposition of Zero-Displacement BC

Note that  $\mathbf{K}$  is singular because  $S_u = \phi$  is assumed.

The displacement BC is imposed by simply getting rid of the columns corresponding to zero-displacements ( $U_i = 0$ ) and the rows corresponding to zero-virtual displacements ( $\delta U_i = 0$ ) in the stiffness matrix  $\mathbf{K}$ .

Ex) When  $U_3 = U_4 = 0$ ,

$$\begin{bmatrix} K_{11} & K_{12} & K_{13} & K_{14} & K_{15} \\ & K_{22} & K_{23} & K_{24} & K_{25} \\ & & K_{33} & K_{34} & K_{35} \\ & \text{sym.} & & K_{44} & K_{45} \\ & & & & K_{55} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \\ U_4 \\ U_5 \end{bmatrix} = \begin{bmatrix} R_1 \\ R_2 \\ R_3 \\ R_4 \\ R_5 \end{bmatrix} \rightarrow \begin{bmatrix} K_{11} & K_{12} & K_{15} \\ & K_{22} & K_{25} \\ \text{sym.} & & K_{55} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_5 \end{bmatrix} = \begin{bmatrix} R_1 \\ R_2 \\ R_5 \end{bmatrix}$$

Then,  $\mathbf{K}_{N \times N}$  is reduced to  $\tilde{\mathbf{K}}_{\tilde{N} \times \tilde{N}}$  ( $\tilde{N} = N - (\# \text{ of prescribed DOFs})$ ) and the displacement and force vectors are also reduced into  $\tilde{\mathbf{U}}$  and  $\tilde{\mathbf{R}}$ . Finally, we get  $\tilde{\mathbf{K}}\tilde{\mathbf{U}} = \tilde{\mathbf{R}}$ . When the displacement BC is properly applied, the equilibrium equation can be solved.

## Properties of Stiffness Matrix

①  $\mathbf{K}$  is a symmetric matrix, i.e.  $\mathbf{K}^T = \mathbf{K}$  or  $K_{ij} = K_{ji}$ .

“Betti’s reciprocal theorem” and “Maxwell’s theorem”

②  $\mathbf{KU} = \mathbf{R}$

External work =  $\frac{1}{2} \mathbf{U}^T \mathbf{R} = \frac{1}{2} \mathbf{U}^T \mathbf{K} \mathbf{U} \geq 0$  (strain energy stored)

When  $\mathbf{U} = \mathbf{0}$  or  $\mathbf{U}$  is the displacement vector corresponding to rigid body motions,

$$\frac{1}{2} \mathbf{U}^T \mathbf{K} \mathbf{U} = 0.$$

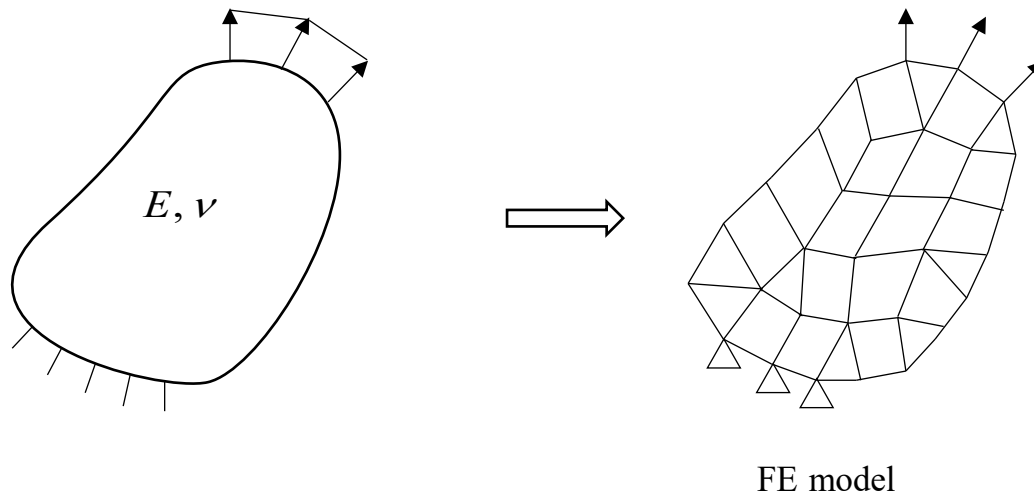
When  $\mathbf{U} \neq \mathbf{0}$ ,  $\frac{1}{2} \tilde{\mathbf{U}}^T \tilde{\mathbf{K}} \tilde{\mathbf{U}} > 0$  for all  $\tilde{\mathbf{U}}$ .  $\rightarrow \tilde{\mathbf{K}}$ : “positive definite” matrix

## FE Solution Procedure

Principal unknown:  $\mathbf{U}$

Step 1) Geometry, material properties, applied load and displacement BC are given.

→ Construct "FE model".



FE model has information on

- nodal positions
- element connectivity (a set of nodes to construct the element)
- material properties ( $E, \nu$ )
- BCs (force & displacement) are only applied at nodes

Step 2) Calculate  $\mathbf{K}^{(m)}$  and  $\mathbf{R}^{(m)}$  of each finite element (element matrix)

Step 3) Assemble  $\mathbf{K}$  and  $\mathbf{R}$  (total matrix).

$$\mathbf{K} = \sum_m \mathbf{K}^{(m)}, \quad \mathbf{R} = \sum_m \mathbf{R}^{(m)}$$

Step 4) Apply the displacement BC.

$$\mathbf{K} \rightarrow \tilde{\mathbf{K}}$$

Step 5) Solve the linear system.

$$\tilde{\mathbf{K}}\tilde{\mathbf{U}} = \tilde{\mathbf{R}} \rightarrow \tilde{\mathbf{U}} = \tilde{\mathbf{K}}^{-1}\tilde{\mathbf{R}} \rightarrow \mathbf{U} \text{ is found.}$$

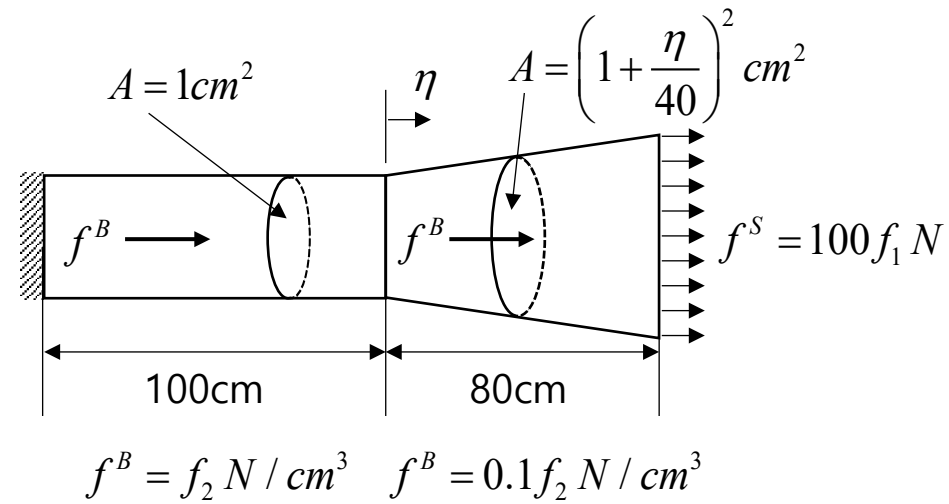
Step 6) Calculate solutions

$$\rightarrow \text{Displacement field of element } m: \mathbf{u}^{(m)} = \mathbf{H}^{(m)}\mathbf{U}$$

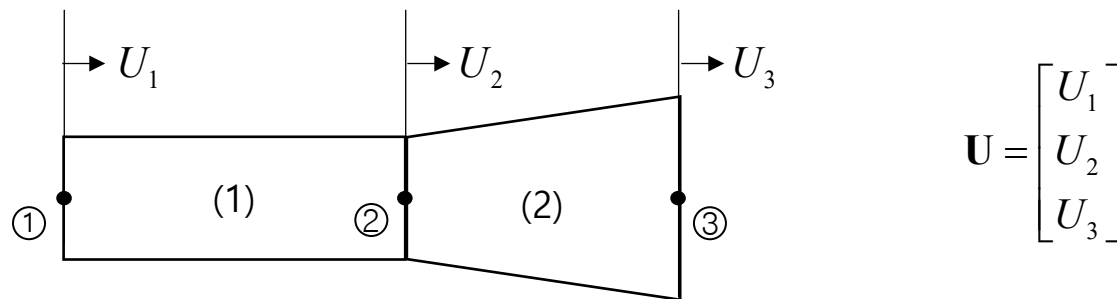
$$\rightarrow \text{Strain field of element } m: \boldsymbol{\varepsilon}^{(m)} = \mathbf{B}^{(m)}\mathbf{U}$$

$$\rightarrow \text{Stress field of element } m: \boldsymbol{\tau}^{(m)} = \mathbf{C}^{(m)}\boldsymbol{\varepsilon}^{(m)} = \mathbf{C}^{(m)}\mathbf{B}^{(m)}\mathbf{U}$$

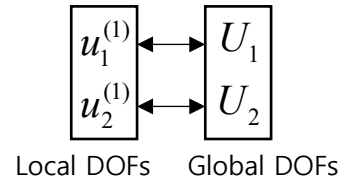
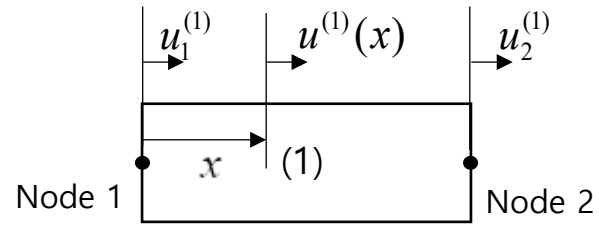
## Example – 1D bar problem



FE model



## Element (1)



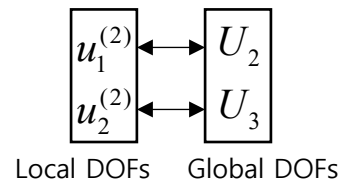
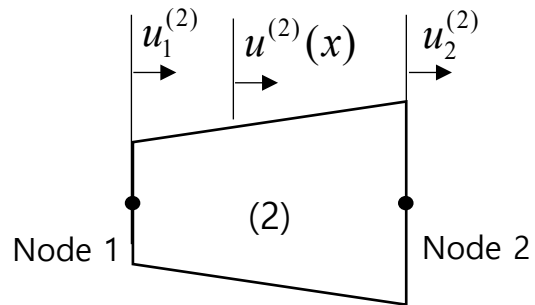
$x$  (locally defined coordinate)

$$u^{(1)}(x) = \left(1 - \frac{x}{100}\right) u_1^{(1)} + \frac{x}{100} u_2^{(1)}$$

$$u^{(1)}(x) = \begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} \end{bmatrix} \begin{bmatrix} u_1^{(1)} \\ u_2^{(1)} \end{bmatrix} = \begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} & 0 \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{H}^{(1)} \mathbf{U}$$

$$\varepsilon_{xx}^{(1)} = \frac{\partial u^{(1)}}{\partial x} = \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} \end{bmatrix} \begin{bmatrix} u_1^{(1)} \\ u_2^{(1)} \end{bmatrix} = \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} & 0 \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{B}^{(1)} \mathbf{U}$$

## Element (2)



$$u^{(2)}(x) = \left(1 - \frac{x}{80}\right) u_1^{(2)} + \frac{x}{80} u_2^{(2)}$$

$$u^{(2)}(x) = \begin{bmatrix} 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \end{bmatrix} = \begin{bmatrix} 0 & 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{H}^{(2)} \mathbf{U}$$

$$\varepsilon_{xx}^{(2)} = \frac{\partial u^{(2)}}{\partial x} = \begin{bmatrix} -\frac{1}{80} & \frac{1}{80} \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \end{bmatrix} = \begin{bmatrix} 0 & -\frac{1}{80} & \frac{1}{80} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{B}^{(2)} \mathbf{U}$$



Stiffness matrix

$$\mathbf{KU} = \mathbf{R} \quad \text{with } \mathbf{U} = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix}$$

$$\begin{aligned} \mathbf{K} &= \sum_{m=1}^2 \mathbf{K}^{(m)} = \mathbf{K}^{(1)} + \mathbf{K}^{(2)} \\ &= \int_{V^{(1)}} \mathbf{B}^{(1)T} E \mathbf{B}^{(1)} dV + \int_{V^{(2)}} \mathbf{B}^{(2)T} E \mathbf{B}^{(2)} dV^{(2)} \end{aligned}$$

$$\mathbf{K} = \int_0^{100} 1 \times \begin{bmatrix} -\frac{1}{100} \\ \frac{1}{100} \\ 0 \end{bmatrix} E \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} & 0 \end{bmatrix} dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 0 \\ -\frac{1}{80} \\ \frac{1}{80} \end{bmatrix} E \begin{bmatrix} 0 & -\frac{1}{80} & \frac{1}{80} \end{bmatrix} dx$$

$$= \frac{E}{100} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} + \frac{13E}{240} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} = \frac{E}{240} \begin{bmatrix} 2.4 & -2.4 & 0 \\ -2.4 & 15.4 & -13 \\ 0 & -13 & 13 \end{bmatrix}$$

## Direct stiffness method

$$\mathbf{K}^{(1)} = \int_0^{100} 1 \times \begin{bmatrix} -\frac{1}{100} \\ \frac{1}{100} \end{bmatrix} E \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} \end{bmatrix} dx = \frac{E}{100} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$\mathbf{K}^{(2)} = \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \times \begin{bmatrix} -\frac{1}{80} \\ \frac{1}{80} \end{bmatrix} E \begin{bmatrix} -\frac{1}{80} & \frac{1}{80} \end{bmatrix} dx = \frac{13E}{240} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$\mathbf{K} = \begin{bmatrix} \frac{E}{100} & -\frac{E}{100} & 0 \\ -\frac{E}{100} & \frac{E}{100} + \frac{13E}{240} & -\frac{13E}{240} \\ 0 & -\frac{13E}{240} & \frac{13E}{240} \end{bmatrix} = \frac{E}{240} \begin{bmatrix} 2.4 & -2.4 & 0 \\ -2.4 & 15.4 & -13 \\ 0 & -13 & 13 \end{bmatrix}$$

Load vector

$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S$$

$$\mathbf{R}_B = \sum_m \mathbf{R}_B^{(m)} \quad \text{with} \quad \mathbf{R}_B^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{(m)} dV^{(m)}$$

$$\mathbf{R}_S = \sum_m \mathbf{R}_S^{(m)} \quad \text{with} \quad \mathbf{R}_S^{(m)} = \int_{S_f^{(m)}} \mathbf{H}_S^{(m)T} \mathbf{f}^{S(m)} dS^{(m)}$$

$$\mathbf{R}_B = \int_0^{100} 1 \times \begin{bmatrix} 1 - \frac{x}{100} \\ \frac{x}{100} \\ 0 \end{bmatrix} f_2 dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 0 \\ 1 - \frac{x}{80} \\ \frac{x}{80} \end{bmatrix} 0.1 f_2 dx = \frac{1}{3} \begin{bmatrix} 150 \\ 186 \\ 68 \end{bmatrix} f_2 \quad (\text{Body force})$$

$$\mathbf{R}_S = \mathbf{R}_S^{(1)} + \mathbf{R}_S^{(2)} = \int_{S_2} \mathbf{H}^{(2)T} \Big|_{x=80} \frac{100f_1}{S_2} dS = \int_{S_2} \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \frac{100f_1}{S_2} dS = \begin{bmatrix} 0 \\ 0 \\ 100f_1 \end{bmatrix} \quad (\text{Surface force})$$

$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S = \begin{bmatrix} 50f_2 \\ 62f_2 \\ \frac{68}{3}f_2 + 100f_1 \end{bmatrix}$$

Equilibrium equation

$$\frac{E}{240} \begin{bmatrix} 2.4 & -2.4 & 0 \\ -2.4 & 15.4 & -13 \\ 0 & -13 & 13 \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \begin{bmatrix} 50f_2 \\ 62f_2 \\ \frac{68}{3}f_2 + 100f_1 \end{bmatrix} \quad (\mathbf{KU} = \mathbf{R}, \mathbf{K} \text{ is singular.})$$

Imposition of displacement BC

$$U_1 = 0 \quad (\text{and } \delta U_1 = 0)$$

$$\frac{E}{240} \begin{bmatrix} 15.4 & -13 \\ -13 & 13 \end{bmatrix} \begin{bmatrix} U_2 \\ U_3 \end{bmatrix} = \begin{bmatrix} 62f_2 \\ \frac{68}{3}f_2 + 100f_1 \end{bmatrix} \quad (\tilde{\mathbf{K}}\tilde{\mathbf{U}} = \tilde{\mathbf{R}})$$

Nodal displacement vector is found:  $\mathbf{U} = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix}$

→ Displacement field:  $\mathbf{u}^{(m)} = \mathbf{H}^{(m)}\mathbf{U}$

→ Strain field:  $\boldsymbol{\varepsilon}_{xx}^{(m)} = \mathbf{B}^{(m)}\mathbf{U}$

→ Stress field:  $\tau_{xx}^{(m)} = E\mathbf{B}^{(m)}\mathbf{U}$

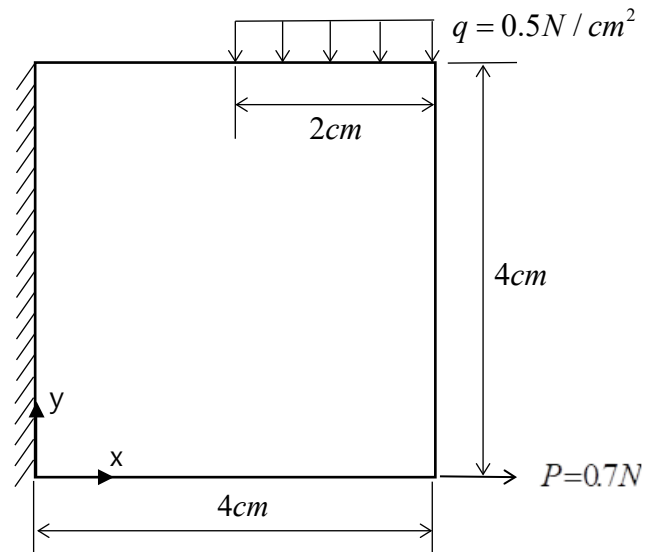
Mass matrix for dynamic analysis

$$\mathbf{M} = \sum_{m=1}^2 \int_{V^{(m)}} \mathbf{H}^{T(m)} \rho^{(m)} \mathbf{H}^{(m)} dV^{(m)}$$

$$\mathbf{M} = \int_0^{100} 1 \times \begin{bmatrix} 1 - \frac{x}{100} \\ \frac{x}{100} \\ 0 \end{bmatrix} \rho \begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} & 0 \end{bmatrix} dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 0 \\ 1 - \frac{x}{80} \\ \frac{x}{80} \end{bmatrix} \rho \begin{bmatrix} 0 & 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix} dx$$

$$\mathbf{M} = \frac{\rho}{6} \begin{bmatrix} 200 & 100 & 0 \\ & 584 & 336 \\ \text{sym.} & & 1024 \end{bmatrix}$$

## Example – 2D plane stress problem



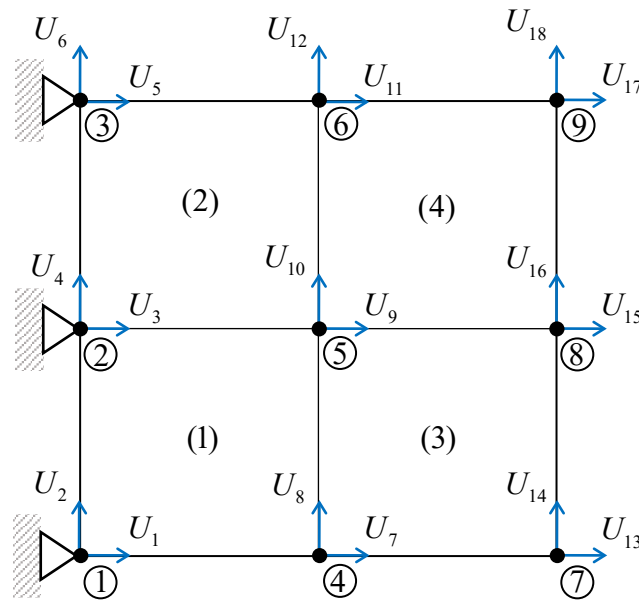
Thickness = 1 ,  $E$  : Young's modulus,  $\nu$  : Poisson's ratio

Plane stress condition

$$\boldsymbol{\tau} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \boldsymbol{\varepsilon} \quad \text{with} \quad \boldsymbol{\tau} = \begin{bmatrix} \tau_{xx} \\ \tau_{yy} \\ \tau_{xy} \end{bmatrix}, \quad \boldsymbol{\varepsilon} = \begin{bmatrix} \varepsilon_{xx} \\ \varepsilon_{yy} \\ \gamma_{xy} \end{bmatrix}.$$



## Finite element model



$$\mathbf{U} = \begin{bmatrix} U_1 \\ U_2 \\ \vdots \\ U_{18} \end{bmatrix}, \quad \mathbf{R} = \begin{bmatrix} R_1 \\ R_2 \\ \vdots \\ R_{18} \end{bmatrix}$$

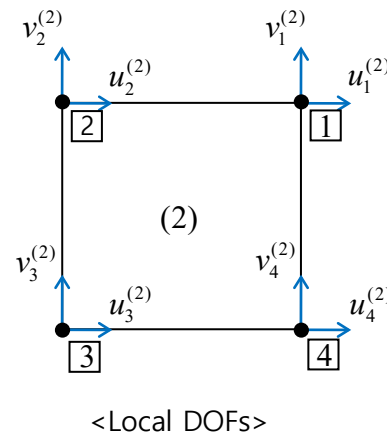
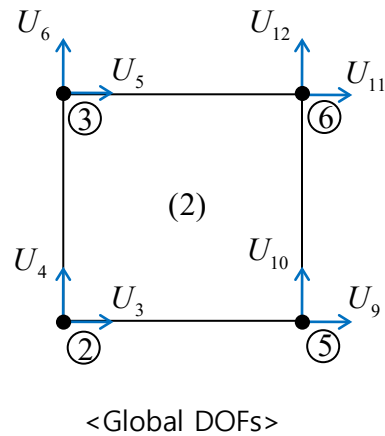
Number of nodes : 9

Number of elements : 4

Number of total DOFs : 18 (9x2)

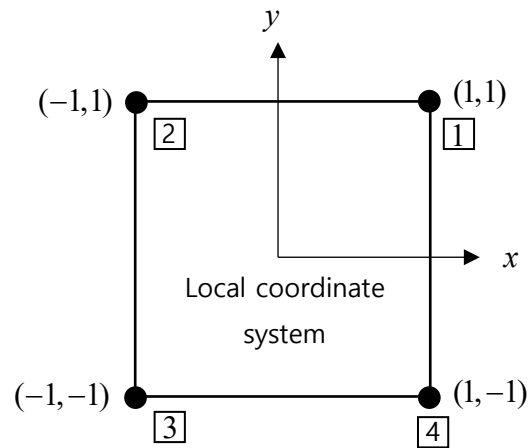
Displacement BC :  $U_1 = U_2 = U_3 = U_4 = U_5 = U_6 = 0$ .

Element stiffness matrices,  $\mathbf{K}^{(m)}$



$$\mathbf{u}^{(2)} = \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \\ u_3^{(2)} \\ u_4^{(2)} \\ v_1^{(2)} \\ v_2^{(2)} \\ v_3^{(2)} \\ v_4^{(2)} \end{bmatrix} \begin{matrix} \leftrightarrow U_{11} \\ \leftrightarrow U_5 \\ \leftrightarrow U_3 \\ \leftrightarrow U_9 \\ \leftrightarrow U_{12} \\ \leftrightarrow U_6 \\ \leftrightarrow U_4 \\ \leftrightarrow U_{10} \end{matrix}$$

$\mathbf{u}^{(2)}$  : Nodal displacement vector of element (2)



Displacement interpolation:

$$\mathbf{u}(x, y) = \begin{bmatrix} u^{(2)}(x, y) \\ v^{(2)}(x, y) \end{bmatrix}$$

$$u^{(2)}(x, y) = \alpha_0 + \alpha_1 x + \alpha_2 y + \alpha_3 xy$$

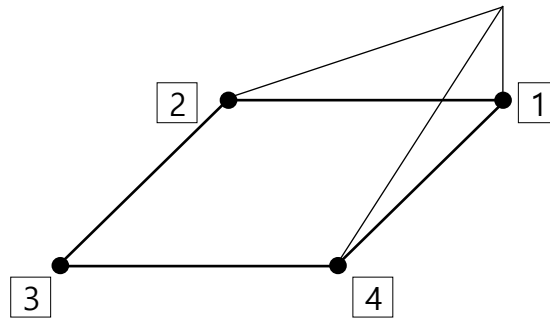
$$u^{(2)}(1,1) = u_1^{(2)}, \quad u^{(2)}(-1,1) = u_2^{(2)}, \quad u^{(2)}(-1,-1) = u_3^{(2)}, \quad u^{(2)}(1,-1) = u_4^{(2)}$$

$$u^{(2)}(x, y) = \sum_i^4 h_i(x, y)u_i^{(2)} = h_1(x, y)u_1^{(2)} + h_2(x, y)u_2^{(2)} + h_3(x, y)u_3^{(2)} + h_4(x, y)u_4^{(2)}$$

with "shape functions"

$$h_1 = \frac{1}{4}(1+x)(1+y), \quad h_2 = \frac{1}{4}(1-x)(1+y), \quad h_3 = \frac{1}{4}(1-x)(1-y), \quad h_4 = \frac{1}{4}(1+x)(1-y)$$

(Note)  $h_i = 1$  at node  $i$ , and  $h_i = 0$  at other nodes.



$$v^{(2)} = \sum_i^4 h_i(x, y)v_i^{(2)} = h_1v_1^{(2)} + h_2v_2^{(2)} + h_3v_3^{(2)} + h_4v_4^{(2)}$$

$$\begin{bmatrix} \mathbf{u}^{(2)} \\ \mathbf{v}^{(2)} \end{bmatrix} = \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \\ u_3^{(2)} \\ u_4^{(2)} \\ v_1^{(2)} \\ v_2^{(2)} \\ v_3^{(2)} \\ v_4^{(2)} \end{bmatrix} = \mathbf{H}^{(2)} \mathbf{u}^{(2)}$$

$$\varepsilon_{xx}^{(2)} = \frac{\partial u^{(2)}}{\partial x} = \begin{bmatrix} \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} & 0 & 0 & 0 & 0 \end{bmatrix} \mathbf{u}^{(2)}$$

$$\varepsilon_{yy}^{(2)} = \frac{\partial v^{(2)}}{\partial y} = \begin{bmatrix} 0 & 0 & 0 & 0 & \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} \end{bmatrix} \mathbf{u}^{(2)}$$

$$\gamma_{xy}^{(2)} = \frac{\partial v^{(2)}}{\partial x} + \frac{\partial u^{(2)}}{\partial y} = \begin{bmatrix} \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} & \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} \end{bmatrix} \mathbf{u}^{(2)}$$

$$\boldsymbol{\varepsilon}^{(2)} = \begin{bmatrix} \varepsilon_{xx}^{(2)} \\ \varepsilon_{yy}^{(2)} \\ \gamma_{xy}^{(2)} \end{bmatrix} = \begin{bmatrix} \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} \\ \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} & \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ \vdots \\ u_4^{(2)} \\ v_1^{(2)} \\ \vdots \\ v_4^{(2)} \end{bmatrix} = \mathbf{B}^{(2)} \mathbf{u}^{(2)}$$

$$\mathbf{K}^{(2)} = \int_{V^{(2)}} \mathbf{B}^{(2)T} \mathbf{C}^{(2)} \mathbf{B}^{(2)} dV^{(2)} = t \int_{-1}^1 \int_{-1}^1 \mathbf{B}^{(2)T}(x, y) \mathbf{C}^{(2)} \mathbf{B}^{(2)}(x, y) dx dy$$

$$\text{with } \mathbf{C}^{(2)} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix}$$

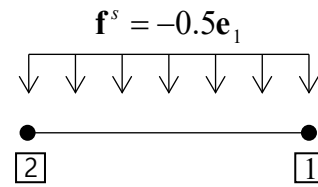
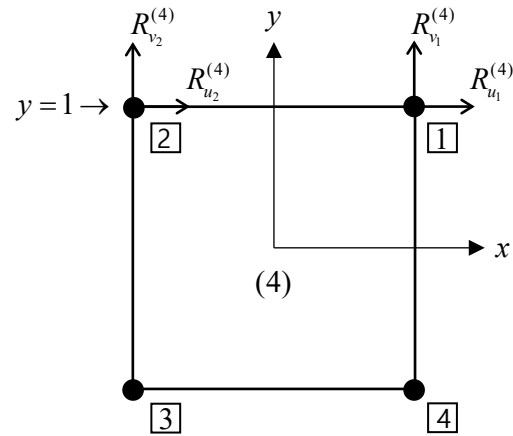
$$\mathbf{K}^{(2)} = \mathbf{K}^{(1)} = \mathbf{K}^{(3)} = \mathbf{K}^{(4)}$$

Stiffness matrix,  $\mathbf{K} : \mathbf{K}^{(1)}_{(8 \times 8)}, \mathbf{K}^{(2)}_{(8 \times 8)}, \mathbf{K}^{(3)}_{(8 \times 8)}, \mathbf{K}^{(4)}_{(8 \times 8)} \rightarrow \mathbf{K}_{(18 \times 18)}$

$$\mathbf{K}_{8 \times 8}^{(2)} = \begin{matrix} & u_1^{(2)} & u_2^{(2)} \\ \delta u_1^{(2)} & \left[ \begin{array}{cc|c} K_{11}^{(2)} & K_{12}^{(2)} & \dots \\ K_{21}^{(2)} & K_{22}^{(2)} & \dots \\ \vdots & \vdots & \ddots \end{array} \right] \\ \delta u_2^{(2)} & & & \\ \vdots & & & \end{matrix}$$

$$\mathbf{K}_{18 \times 18} = \begin{matrix} & U_1 & U_2 & \dots & U_5 & \dots & U_{11} & \dots & U_{18} \\ \delta U_1 & & & & | & & | & & \\ \delta U_2 & & & & | & & | & & \\ \vdots & & & & | & & | & & \\ \delta U_{11} & - & - & - & K_{12}^{(2)} & - & K_{11}^{(2)} & - & - & - \\ \vdots & & & & | & & | & & \\ \delta U_{18} & & & & | & & | & & \end{matrix}$$

Load vector:  $\mathbf{R}$



$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S$$

$$\mathbf{R}_B = \sum_m \mathbf{R}_B^{(m)} \quad \text{with} \quad \mathbf{R}_B^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{(m)} dV^{(m)}$$

$$\mathbf{R}_S = \sum_m \mathbf{R}_S^{(m)} \quad \text{with} \quad \mathbf{R}_S^{(m)} = \int_{S_f^{(m)}} \mathbf{H}_S^{(m)T} \mathbf{f}^{S(m)} dS^{(m)}$$



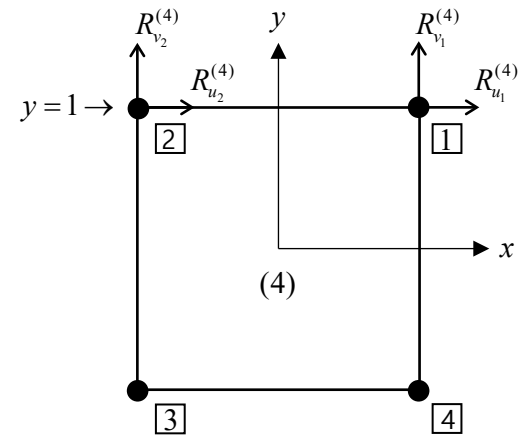
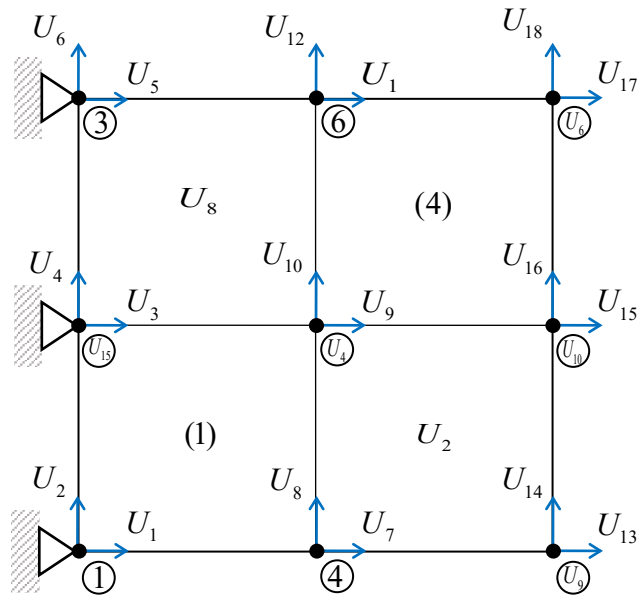
$$\begin{bmatrix} \mathbf{u}^{(4)} \\ \mathbf{v}^{(4)} \end{bmatrix} = \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \mathbf{u}^{(4)} = \mathbf{H}^{(4)} \mathbf{u}^{(4)}$$

$$\mathbf{H}_S^{(4)} = \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \text{ at } y=1$$

$$= \begin{bmatrix} (1+x)/2 & (1-x)/2 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & (1+x)/2 & (1-x)/2 & 0 & 0 \end{bmatrix}$$

$$\mathbf{f}^{S(4)} = \begin{bmatrix} 0 \\ -0.5 \end{bmatrix}$$

$$\mathbf{R}^{(4)} = \mathbf{R}_S^{(4)} = \int_{S_f^{(4)}} \mathbf{H}_S^{(4)T} \mathbf{f}^{S(4)} dS^{(4)} = 1 \times \int_{-1}^1 \mathbf{H}_S^{(4)T} \mathbf{f}^{S(4)} dx = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ -0.5 \\ -0.5 \\ 0 \\ 0 \end{bmatrix}$$



$$\mathbf{R} = \begin{bmatrix} 0 \\ \vdots \\ 0 \\ -0.5 \leftarrow R_{12} \\ 0.7 \leftarrow R_{13} \\ \vdots \\ 0 \\ -0.5 \leftarrow R_{18} \end{bmatrix}$$

We obtain the equilibrium equation.

$$\mathbf{KU} = \mathbf{R}$$

Imposition of displacement BC

$$\text{Displacement BC : } U_1 = U_2 = U_3 = U_4 = U_5 = U_6 = 0$$

$$\underset{12 \times 12}{\tilde{\mathbf{K}}} \tilde{\mathbf{U}} = \tilde{\mathbf{R}}$$

Strain and stress

$$\begin{cases} \boldsymbol{\varepsilon}^{(m)} = \mathbf{B}^{(m)} \mathbf{u}^{(m)} \\ \boldsymbol{\tau}^{(m)} = \mathbf{C}^{(m)} \mathbf{B}^{(m)} \mathbf{u}^{(m)} \end{cases}$$