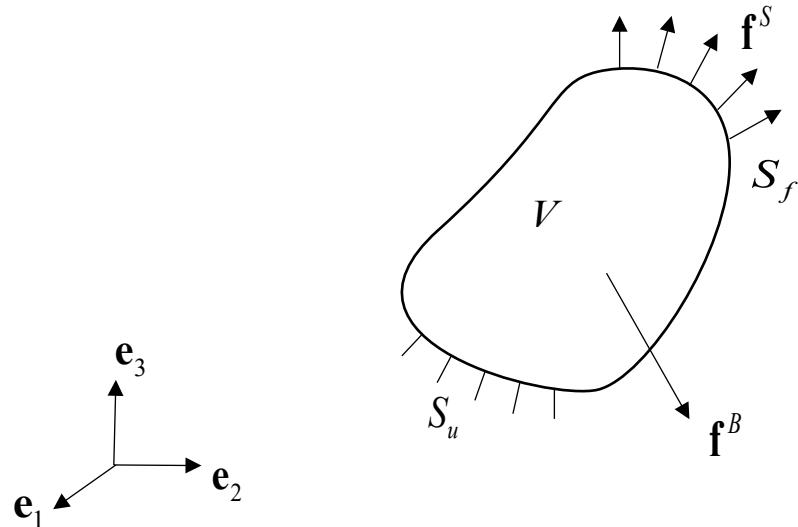


## 4. Finite Element Formulation



$$\text{PVW: } \int_V \tau_{ij} \delta \varepsilon_{ij} dV = \int_V f_i^B \delta u_i dV + \int_{S_f} f_i^S \delta u_i dS$$

Let us assume

- ① Small displacement:  $V_o \approx V$
- ② Linear elastic material:  $\tau_{ij} = C_{ijkl} \varepsilon_{kl}$

In the finite element formulation,

$$\boldsymbol{\tau} = \begin{bmatrix} \tau_{11} \\ \tau_{22} \\ \tau_{33} \\ \tau_{12} \\ \tau_{23} \\ \tau_{31} \end{bmatrix}, \quad \boldsymbol{\varepsilon} = \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ \gamma_{12} \\ \gamma_{23} \\ \gamma_{31} \end{bmatrix},$$

where  $\gamma_{ij} = 2\varepsilon_{ij}$  ( $i \neq j$ ) : engineering shear strain

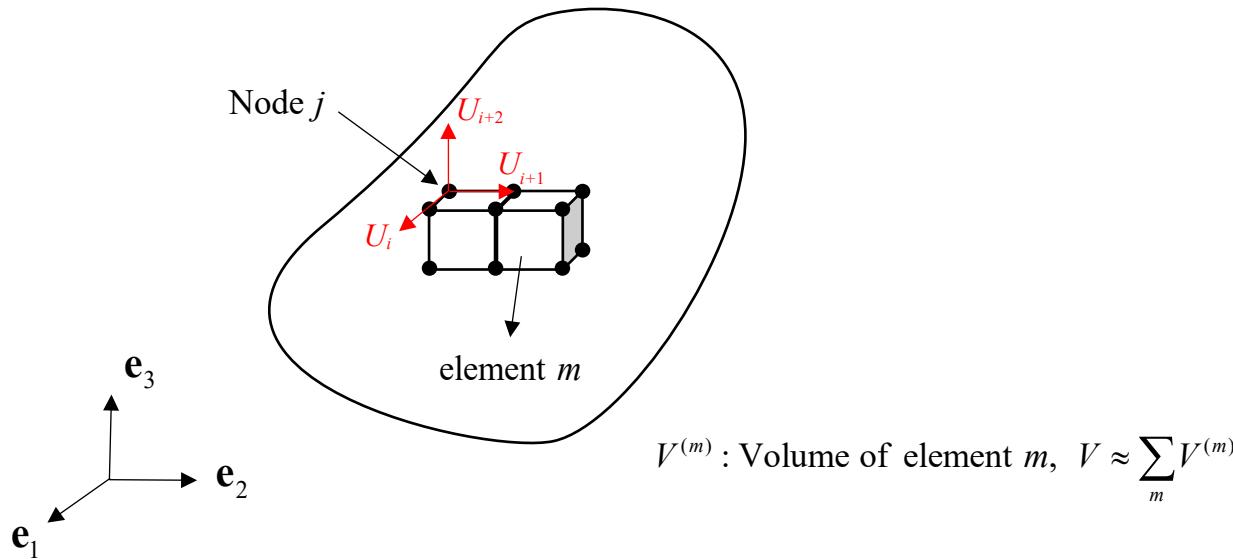
$$\tau_{ij}\varepsilon_{ij} = \sum_{i=1}^3 \sum_{j=1}^3 \tau_{ij}\varepsilon_{ij} = \boldsymbol{\varepsilon}^T \boldsymbol{\tau} = \boldsymbol{\varepsilon} \cdot \boldsymbol{\tau} \quad (\text{note: } \tau_{12}\varepsilon_{12} + \tau_{21}\varepsilon_{21} = 2\tau_{12}\varepsilon_{12} = \tau_{12}\gamma_{12})$$

Material Law:  $\boldsymbol{\tau} = \mathbf{C}\boldsymbol{\varepsilon}$

PVW in vector/matrix form:

$$\int_V \delta \boldsymbol{\varepsilon}^T \mathbf{C} \boldsymbol{\varepsilon} dV = \int_V \delta \mathbf{u}^T \mathbf{f}^B dV + \int_{S_f} \delta \mathbf{u}^T \mathbf{f}^S dS$$

## Finite Element Discretization



Let us assume that  $S_f = S$  and  $S_u = \phi$   $\rightarrow$  "PVW still works."

Nodal displacement at node  $j$   $\rightarrow$   $\begin{bmatrix} U_i \\ U_{i+1} \\ U_{i+2} \end{bmatrix}$

Nodal displacement vector (nodal DOFs vector),

$$\mathbf{U} = \begin{bmatrix} U_1 \\ \vdots \\ U_i \\ U_{i+1} \\ \vdots \\ U_N \end{bmatrix}, \quad N: \text{number of the total DOFs}$$

$$\text{PVW: } \sum_m \int_{V^{(m)}} \delta \boldsymbol{\varepsilon}^{(m)\top} \mathbf{C}^{(m)} \boldsymbol{\varepsilon}^{(m)} dV^{(m)} = \sum_m \int_{V^{(m)}} \delta \mathbf{u}^{(m)\top} \mathbf{f}^{B(m)} dV^{(m)} + \sum_m \int_{S_1^{(m)} \dots S_q^{(m)}} \delta \mathbf{u}^{s(m)\top} \mathbf{f}^{S(m)} dS^{(m)},$$

where  $S_1^{(m)} \dots S_q^{(m)}$  are surfaces of "element  $m$ " on boundary.

## Displacement Interpolations

$$\mathbf{u}^{(m)} = \mathbf{H}^{(m)} \mathbf{U} \text{ (interpolation of displacement)}$$

$$\delta\mathbf{u}^{(m)} = \mathbf{H}^{(m)} \delta\mathbf{U} \text{ (interpolation of virtual displacement)}$$

where

$\mathbf{u}^{(m)}$  : displacement field of element  $m$

$\delta\mathbf{u}^{(m)}$  : virtual displacement field of element  $m$

$\mathbf{U}$  : nodal displacement vector

$\delta\mathbf{U}$  : virtual nodal displacement vector

$\mathbf{H}^{(m)}$  : displacement interpolation matrix for element  $m$

(Note) The same interpolation is used for real and virtual displacements.

→ "symmetric stiffness matrix"

From the strain-displacement relation,

$$\varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \quad \rightarrow \quad \boldsymbol{\varepsilon}^{(m)} = \mathbf{B}^{(m)} \mathbf{U}$$

$$\delta \varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial \delta u_i}{\partial x_j} + \frac{\partial \delta u_j}{\partial x_i} \right) \quad \rightarrow \quad \delta \boldsymbol{\varepsilon}^{(m)} = \mathbf{B}^{(m)} \delta \mathbf{U}$$

$\boldsymbol{\varepsilon}^{(m)}$  : strain field of element m

$\delta \boldsymbol{\varepsilon}^{(m)}$  : virtual strain field of element m

$\mathbf{B}^{(m)}$  : strain interpolation matrix for element m

Using the displacement and strain interpolations in PVW, the following equation is obtained

$$\delta \mathbf{U}^T \left[ \sum_m \int_{V^{(m)}} \mathbf{B}^{(m)\top} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)} \right] \mathbf{U} = \delta \mathbf{U}^T \left[ \sum_m \int_{V^{(m)}} \mathbf{H}^{(m)\top} \mathbf{f}^{B(m)} dV^{(m)} + \sum_m \int_{S_1^{(m)} \dots S_q^{(m)}} \mathbf{H}_s^{(m)\top} \mathbf{f}^{S(m)} dS^{(m)} \right] \quad (\text{eq. 3.1})$$

Finally, we obtain a set of  $N$  linear equations.  $\rightarrow$  "static equilibrium equations"

$$\boxed{\mathbf{K}\mathbf{U} = \mathbf{R}},$$

where  $\mathbf{K} = \sum_m \mathbf{K}^{(m)}$  with  $\mathbf{K}^{(m)} = \int_{V^{(m)}} \mathbf{B}^{(m)\top} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)}$

$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S \quad \begin{cases} \mathbf{R}_B = \sum_m \mathbf{R}_B^{(m)} \quad \text{with } \mathbf{R}_B^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)\top} \mathbf{f}^{B(m)} dV^{(m)} \\ \mathbf{R}_S = \sum_m \mathbf{R}_S^{(m)} \quad \text{with } \mathbf{R}_S^{(m)} = \int_{S_1^{(m)} \dots S_q^{(m)}} \mathbf{H}_s^{(m)\top} \mathbf{f}^{s(m)} dS^{(m)} \end{cases}$$

**K:** Stiffness matrix

**U:** Nodal displacement vector

**R:** Nodal force vector

## Imposition of Zero-Displacement BC

Note that  $\mathbf{K}$  is singular because  $S_u = \phi$  is assumed.

The displacement BC is imposed by simply getting rid of the columns corresponding to zero-displacements ( $U_i = 0$ ) and the rows corresponding to zero-virtual displacements ( $\delta U_i = 0$ ) in the stiffness matrix  $\mathbf{K}$ .

Ex) When  $U_3 = U_4 = 0$ ,

$$\begin{bmatrix} K_{11} & K_{12} & \cancel{K_{13}} & \cancel{K_{14}} & K_{15} \\ K_{21} & K_{22} & \cancel{K_{23}} & \cancel{K_{24}} & K_{25} \\ \hline \cancel{K_{31}} & \cancel{K_{32}} & K_{33} & K_{34} & K_{35} \\ \hline \text{sym.} & K_{41} & K_{42} & \cancel{K_{43}} & \cancel{K_{44}} \\ & & & K_{45} & K_{55} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \\ U_4 \\ U_5 \end{bmatrix} = \begin{bmatrix} R_1 \\ R_2 \\ R_3 \\ R_4 \\ R_5 \end{bmatrix} \rightarrow \begin{bmatrix} K_{11} & K_{12} & K_{15} \\ K_{21} & K_{22} & K_{25} \\ \hline \text{sym.} & K_{55} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_5 \end{bmatrix} = \begin{bmatrix} R_1 \\ R_2 \\ R_5 \end{bmatrix}$$

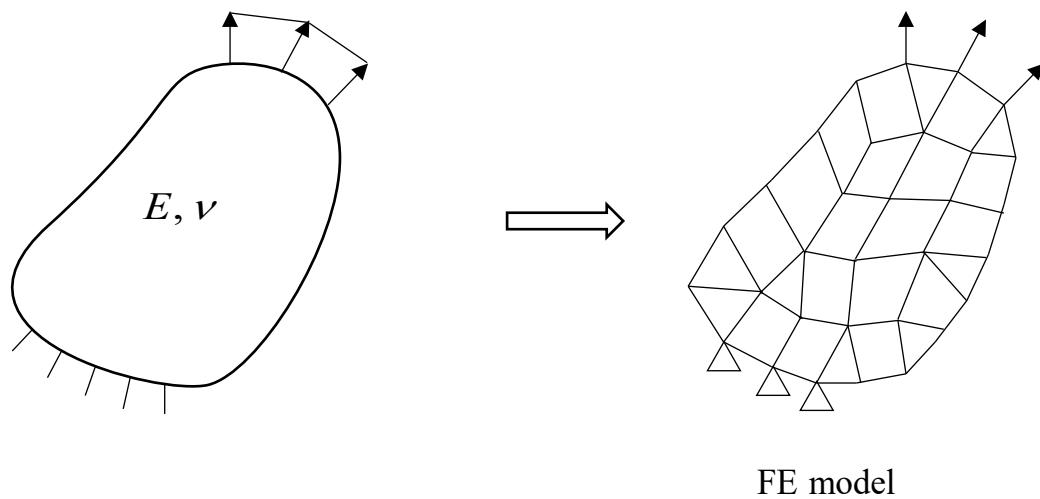
Then,  $\mathbf{K}_{N \times N}$  is reduced to  $\tilde{\mathbf{K}}_{\tilde{N} \times \tilde{N}}$  ( $\tilde{N} = N - (\# \text{ of prescribed DOFs})$ ) and the displacement and force vectors are also reduced into  $\tilde{\mathbf{U}}$  and  $\tilde{\mathbf{R}}$ . Finally, we get  $\tilde{\mathbf{K}}\tilde{\mathbf{U}} = \tilde{\mathbf{R}}$ . When the displacement BC is properly applied, the equilibrium equation can be solved.

## FE Solution Procedure

Principal unknown:  $\mathbf{U}$

Step 1) Geometry, material properties, applied load and displacement BC are given.

→ Construct "FE model".



FE model has information on

- nodal positions
- element connectivity (a set of nodes to construct the element)
- material properties ( $E, \nu$ )
- BCs (force & displacement) are only applied at nodes

Step 2) Calculate  $\mathbf{K}^{(m)}$  and  $\mathbf{R}^{(m)}$  of each finite element (element matrix)

Step 3) Assemble  $\mathbf{K}$  and  $\mathbf{R}$  (total matrix).

$$\mathbf{K} = \sum_m \mathbf{K}^{(m)}, \quad \mathbf{R} = \sum_m \mathbf{R}^{(m)}$$

Step 4) Apply the displacement BC.

$$\mathbf{K} \rightarrow \tilde{\mathbf{K}}$$

Step 5) Solve the linear system.

$$\tilde{\mathbf{K}}\tilde{\mathbf{U}} = \tilde{\mathbf{R}} \rightarrow \tilde{\mathbf{U}} = \tilde{\mathbf{K}}^{-1}\tilde{\mathbf{R}} \quad \rightarrow \quad \mathbf{U} \text{ is found.}$$

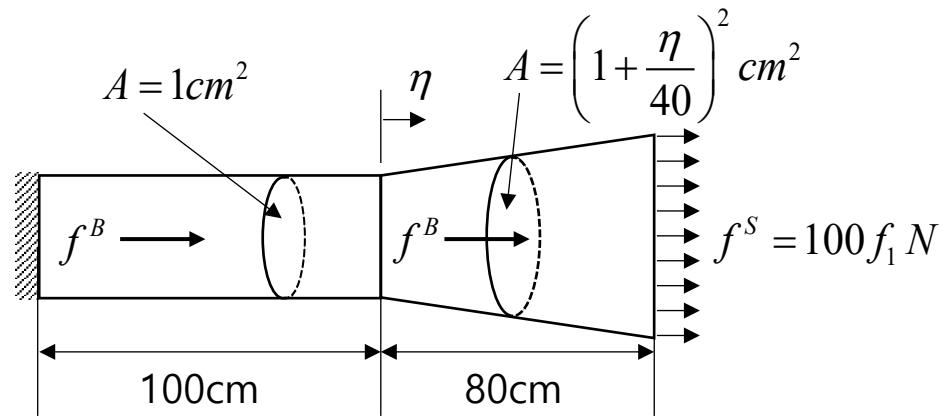
Step 6) Calculate solutions

$$\rightarrow \text{Displacement field of element } m: \mathbf{u}^{(m)} = \mathbf{H}^{(m)}\mathbf{U}$$

$$\rightarrow \text{Strain field of element } m: \boldsymbol{\epsilon}^{(m)} = \mathbf{B}^{(m)}\mathbf{U}$$

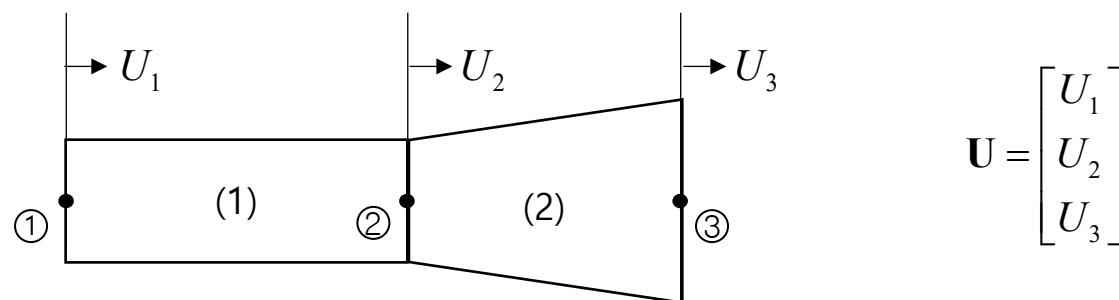
→ Stress field of element  $m$ :  $\boldsymbol{\tau}^{(m)} = \mathbf{C}^{(m)} \boldsymbol{\epsilon}^{(m)} = \mathbf{C}^{(m)} \mathbf{B}^{(m)} \mathbf{U}$

## Example – 1D bar problem

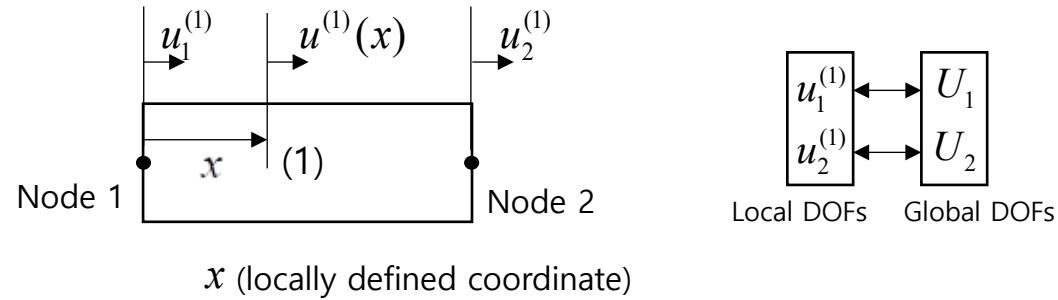


$$f^B = f_2 \text{ N/cm}^3 \quad f^S = 0.1 f_2 \text{ N/cm}^3$$

FE model



## Element (1)

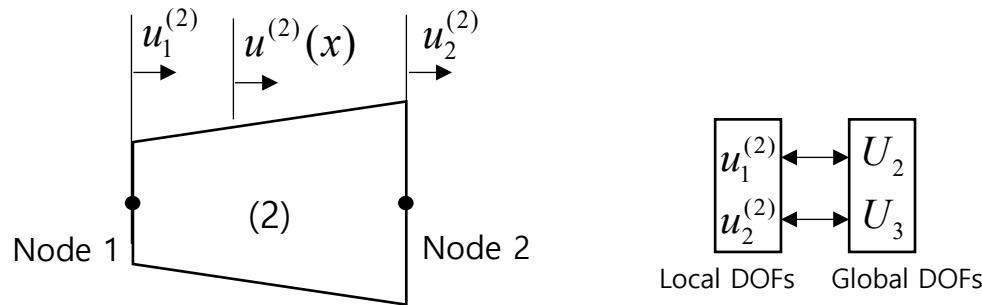


$$u^{(1)}(x) = \left(1 - \frac{x}{100}\right)u_1^{(1)} + \frac{x}{100}u_2^{(1)}$$

$$u^{(1)}(x) = \begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} \end{bmatrix} \begin{bmatrix} u_1^{(1)} \\ u_2^{(1)} \end{bmatrix} = \begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} & 0 \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{H}^{(1)} \mathbf{U}$$

$$\boldsymbol{\varepsilon}_{xx}^{(1)} = \frac{\partial u^{(1)}}{\partial x} = \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} \end{bmatrix} \begin{bmatrix} u_1^{(1)} \\ u_2^{(1)} \end{bmatrix} = \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} & 0 \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{B}^{(1)} \mathbf{U}$$

## Element (2)



$$u^{(2)}(x) = \left(1 - \frac{x}{80}\right)u_1^{(2)} + \frac{x}{80}u_2^{(2)}$$

$$u^{(2)}(x) = \begin{bmatrix} 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \end{bmatrix} = \begin{bmatrix} 0 & 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{H}^{(2)} \mathbf{U}$$

$$\boldsymbol{\varepsilon}_{xx}^{(2)} = \frac{\partial u^{(2)}}{\partial x} = \begin{bmatrix} -\frac{1}{80} & \frac{1}{80} \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \end{bmatrix} = \begin{bmatrix} 0 & -\frac{1}{80} & \frac{1}{80} \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \mathbf{B}^{(2)} \mathbf{U}$$

Stiffness matrix

$$\mathbf{KU} = \mathbf{R} \quad \text{with } \mathbf{U} = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix}$$

$$\mathbf{K} = \sum_{m=1}^2 \mathbf{K}^{(m)} = \mathbf{K}^{(1)} + \mathbf{K}^{(2)}$$

$$= \int_{V^{(1)}} \mathbf{B}^{(1)T} E \mathbf{B}^{(1)} dV + \int_{V^{(2)}} \mathbf{B}^{(2)T} E \mathbf{B}^{(2)} dV^{(2)}$$

$$\mathbf{K} = \int_0^{100} 1 \times \begin{bmatrix} -\frac{1}{100} \\ \frac{1}{100} \\ 0 \end{bmatrix} E \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} & 0 \end{bmatrix} dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 0 \\ -\frac{1}{80} \\ \frac{1}{80} \end{bmatrix} E \begin{bmatrix} 0 & -\frac{1}{80} & \frac{1}{80} \end{bmatrix} dx$$

$$= \frac{E}{100} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} + \frac{13E}{240} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} = \frac{E}{240} \begin{bmatrix} 2.4 & -2.4 & 0 \\ -2.4 & 15.4 & -13 \\ 0 & -13 & 13 \end{bmatrix}$$

Direct stiffness method

$$\mathbf{K}^{(1)} = \int_0^{100} 1 \times \begin{bmatrix} -\frac{1}{100} \\ -\frac{1}{100} \\ \frac{1}{100} \end{bmatrix} E \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} \end{bmatrix} dx = \frac{E}{100} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$\mathbf{K}^{(2)} = \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \times \begin{bmatrix} -\frac{1}{80} \\ \frac{1}{80} \end{bmatrix} E \begin{bmatrix} -\frac{1}{80} & \frac{1}{80} \end{bmatrix} dx = \frac{13E}{240} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

$$\mathbf{K} = \begin{bmatrix} \frac{E}{100} & -\frac{E}{100} & 0 \\ -\frac{E}{100} & \frac{E}{100} + \frac{13E}{240} & \frac{-13E}{240} \\ 0 & \frac{-13E}{240} & \frac{13E}{240} \end{bmatrix} = \frac{E}{240} \begin{bmatrix} 2.4 & -2.4 & 0 \\ -2.4 & 15.4 & -13 \\ 0 & -13 & 13 \end{bmatrix}$$

Load vector

$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S$$

$$\mathbf{R}_B = \sum_m \mathbf{R}_B^{(m)} \quad \text{with} \quad \mathbf{R}_B^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{(m)} dV^{(m)}$$

$$\mathbf{R}_S = \sum_m \mathbf{R}_S^{(m)} \quad \text{with} \quad \mathbf{R}_S^{(m)} = \int_{S_f^{(m)}} \mathbf{H}_S^{(m)T} \mathbf{f}^{S(m)} dS^{(m)}$$

$$\mathbf{R}_B = \int_0^{100} 1 \times \begin{bmatrix} 1 - \frac{x}{100} \\ \frac{x}{100} \\ 0 \end{bmatrix} f_2 dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 0 \\ 1 - \frac{x}{80} \\ \frac{x}{80} \end{bmatrix} 0.1 f_2 dx = \frac{1}{3} \begin{bmatrix} 150 \\ 186 \\ 68 \end{bmatrix} f_2 \quad (\text{Body force})$$

$$\mathbf{R}_S = \mathbf{R}_S^{(1)} + \mathbf{R}_S^{(2)} = \int_{S_2} \mathbf{H}^{(2)T} \Big|_{x=80} \frac{100f_1}{S_2} dS = \int_{S_2} \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \frac{100f_1}{S_2} dS = \begin{bmatrix} 0 \\ 0 \\ 100f_1 \end{bmatrix} \quad (\text{Surface force})$$

$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S = \begin{bmatrix} 50f_2 \\ 62f_2 \\ \frac{68}{3}f_2 + 100f_1 \end{bmatrix}$$

Equilibrium equation

$$\frac{E}{240} \begin{bmatrix} 2.4 & -2.4 & 0 \\ -2.4 & 15.4 & -13 \\ 0 & -13 & 13 \end{bmatrix} \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} = \begin{bmatrix} 50f_2 \\ 62f_2 \\ \frac{68}{3}f_2 + 100f_1 \end{bmatrix} \quad (\mathbf{KU} = \mathbf{R}, \mathbf{K} \text{ is singular.})$$

## Imposition of displacement BC

$$U_1 = 0 \quad (\text{and } \delta U_1 = 0)$$

$$\frac{E}{240} \begin{bmatrix} 15.4 & -13 \\ -13 & 13 \end{bmatrix} \begin{bmatrix} U_2 \\ U_3 \end{bmatrix} = \begin{bmatrix} 62f_2 \\ \frac{68}{3}f_2 + 100f_1 \end{bmatrix} \quad (\tilde{\mathbf{K}}\tilde{\mathbf{U}} = \tilde{\mathbf{R}})$$

Nodal displacement vector is found:  $\mathbf{U} = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix}$

→ Displacement field:  $\mathbf{u}^{(m)} = \mathbf{H}^{(m)}\mathbf{U}$

→ Strain field:  $\boldsymbol{\varepsilon}_{xx}^{(m)} = \mathbf{B}^{(m)}\mathbf{U}$

→ Stress field:  $\boldsymbol{\tau}_{xx}^{(m)} = E\mathbf{B}^{(m)}\mathbf{U}$

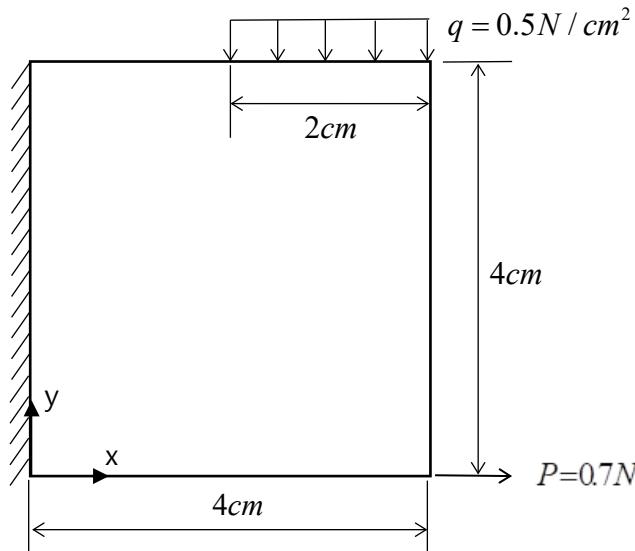
Mass matrix for dynamic analysis

$$\mathbf{M} = \sum_{m=1}^2 \int_{V^{(m)}} \mathbf{H}^{T(m)} \rho^{(m)} \mathbf{H}^{(m)} dV^{(m)}$$

$$\mathbf{M} = \int_0^{100} 1 \times \begin{bmatrix} 1 - \frac{x}{100} \\ \frac{x}{100} \\ 0 \end{bmatrix} \rho \begin{bmatrix} 1 - \frac{x}{100} & \frac{x}{100} & 0 \end{bmatrix} dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 0 \\ 1 - \frac{x}{80} \\ \frac{x}{80} \end{bmatrix} \rho \begin{bmatrix} 0 & 1 - \frac{x}{80} & \frac{x}{80} \end{bmatrix} dx$$

$$\mathbf{M} = \frac{\rho}{6} \begin{bmatrix} 200 & 100 & 0 \\ & 584 & 336 \\ \text{sym.} & & 1024 \end{bmatrix}$$

## Example – 2D plane stress problem

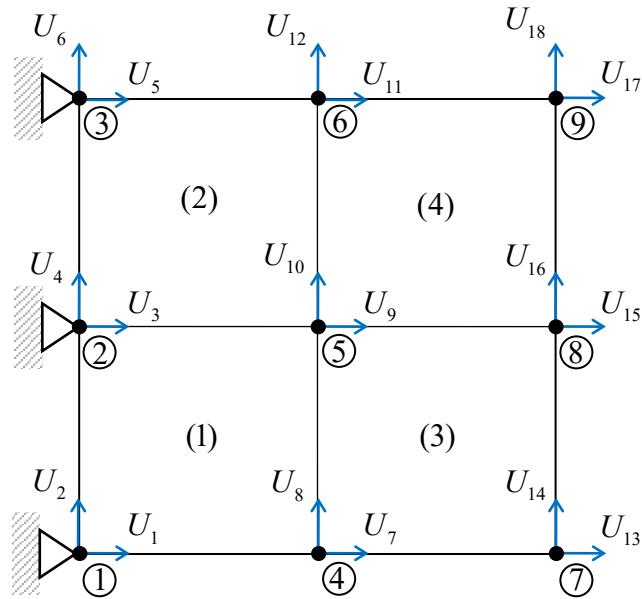


Thickness =1 ,  $E$  : Young's modulus,  $\nu$  : Poisson's ratio

Plane stress condition

$$\boldsymbol{\tau} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \boldsymbol{\epsilon} \quad \text{with} \quad \boldsymbol{\tau} = \begin{bmatrix} \tau_{xx} \\ \tau_{yy} \\ \tau_{xy} \end{bmatrix}, \quad \boldsymbol{\epsilon} = \begin{bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \gamma_{xy} \end{bmatrix}.$$

## Finite element model



$$\mathbf{U} = \begin{bmatrix} U_1 \\ U_2 \\ \vdots \\ U_{18} \end{bmatrix}, \quad \mathbf{R} = \begin{bmatrix} R_1 \\ R_2 \\ \vdots \\ R_{18} \end{bmatrix}$$

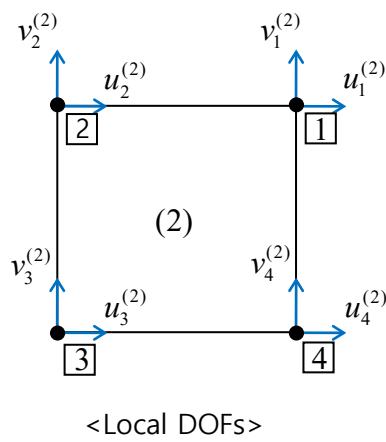
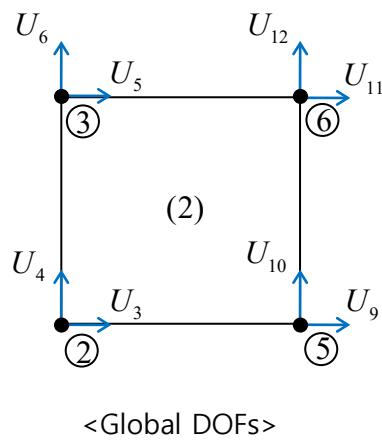
Number of nodes : 9

Number of elements : 4

Number of total DOFs : 18 (9x2)

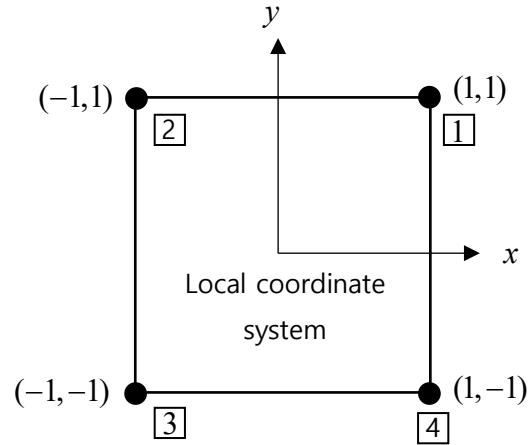
Displacement BC :  $U_1 = U_2 = U_3 = U_4 = U_5 = U_6 = 0$ .

## Element stiffness matrices, $\mathbf{K}^{(m)}$



$$\mathbf{u}^{(2)} = \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \\ u_3^{(2)} \\ u_4^{(2)} \\ v_1^{(2)} \\ v_2^{(2)} \\ v_3^{(2)} \\ v_4^{(2)} \end{bmatrix} \leftrightarrow \begin{array}{l} U_{11} \\ U_5 \\ U_3 \\ U_9 \\ U_{12} \\ U_6 \\ U_4 \\ U_{10} \end{array}$$

$\mathbf{u}^{(2)}$  : Nodal displacement vector of element (2)



Displacement interpolation:

$$\mathbf{u}(x, y) = \begin{bmatrix} u^{(2)}(x, y) \\ v^{(2)}(x, y) \end{bmatrix}$$

$$u^{(2)}(x, y) = \alpha_0 + \alpha_1 x + \alpha_2 y + \alpha_3 xy$$

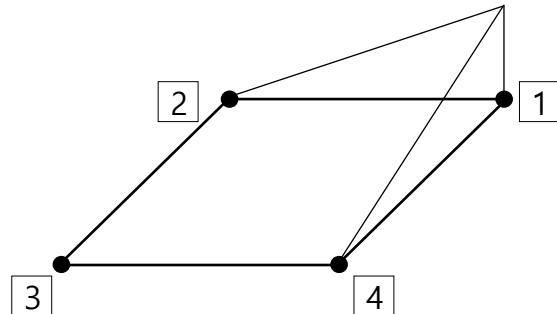
$$u^{(2)}(1,1) = u_1^{(2)}, \quad u^{(2)}(-1,1) = u_2^{(2)}, \quad u^{(2)}(-1,-1) = u_3^{(2)}, \quad u^{(2)}(1,-1) = u_4^{(2)}$$

$$u^{(2)}(x, y) = \sum_i^4 h_i(x, y) u_i^{(2)} = h_1(x, y) u_1^{(2)} + h_2(x, y) u_2^{(2)} + h_3(x, y) u_3^{(2)} + h_4(x, y) u_4^{(2)}$$

with "shape functions"

$$h_1 = \frac{1}{4}(1+x)(1+y), \quad h_2 = \frac{1}{4}(1-x)(1+y), \quad h_3 = \frac{1}{4}(1-x)(1-y), \quad h_4 = \frac{1}{4}(1+x)(1-y)$$

(Note)  $h_i = 1$  at node  $i$ , and  $h_i = 0$  at other nodes.



$$v^{(2)} = \sum_i^4 h_i(x, y) v_i^{(2)} = h_1 v_1^{(2)} + h_2 v_2^{(2)} + h_3 v_3^{(2)} + h_4 v_4^{(2)}$$

$$\begin{bmatrix} u^{(2)} \\ v^{(2)} \end{bmatrix} = \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ u_2^{(2)} \\ u_3^{(2)} \\ u_4^{(2)} \\ v_1^{(2)} \\ v_2^{(2)} \\ v_3^{(2)} \\ v_4^{(2)} \end{bmatrix} = \mathbf{H}^{(2)} \mathbf{u}^{(2)}$$

$$\varepsilon_{xx}^{(2)} = \frac{\partial u^{(2)}}{\partial x} = \begin{bmatrix} \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} & 0 & 0 & 0 & 0 \end{bmatrix} \mathbf{u}^{(2)}$$

$$\varepsilon_{yy}^{(2)} = \frac{\partial v^{(2)}}{\partial y} = \begin{bmatrix} 0 & 0 & 0 & 0 & \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} \end{bmatrix} \mathbf{u}^{(2)}$$

$$\gamma_{xy}^{(2)} = \frac{\partial v^{(2)}}{\partial x} + \frac{\partial u^{(2)}}{\partial y} = \begin{bmatrix} \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} & \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} \end{bmatrix} \mathbf{u}^{(2)}$$

$$\boldsymbol{\varepsilon}^{(2)} = \begin{bmatrix} \varepsilon_{xx}^{(2)} \\ \varepsilon_{yy}^{(2)} \\ \gamma_{xy}^{(2)} \end{bmatrix} = \begin{bmatrix} \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} \\ \frac{\partial h_1}{\partial y} & \frac{\partial h_2}{\partial y} & \frac{\partial h_3}{\partial y} & \frac{\partial h_4}{\partial y} & \frac{\partial h_1}{\partial x} & \frac{\partial h_2}{\partial x} & \frac{\partial h_3}{\partial x} & \frac{\partial h_4}{\partial x} \end{bmatrix} \begin{bmatrix} u_1^{(2)} \\ \vdots \\ u_4^{(2)} \\ v_1^{(2)} \\ \vdots \\ v_4^{(2)} \end{bmatrix} = \mathbf{B}^{(2)} \mathbf{u}^{(2)}$$

$$\mathbf{K}^{(2)} = \int_{V^{(2)}} \mathbf{B}^{(2)T} \mathbf{C}^{(2)} \mathbf{B}^{(2)} dV^{(2)} = t \int_{-1}^1 \int_{-1}^1 \mathbf{B}^{(2)T}(x, y) \mathbf{C}^{(2)} \mathbf{B}^{(2)}(x, y) dx dy$$

with  $\mathbf{C}^{(2)} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix}$

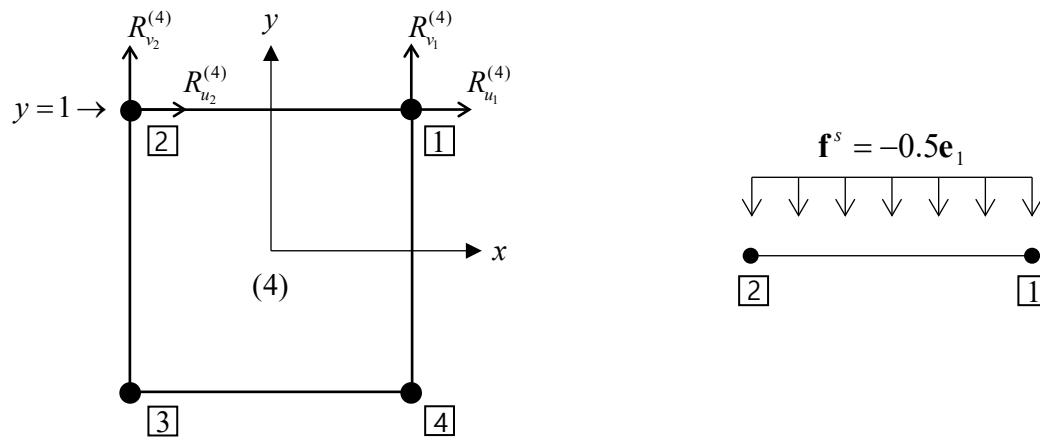
$$\mathbf{K}^{(2)} = \mathbf{K}^{(1)} = \mathbf{K}^{(3)} = \mathbf{K}^{(4)}$$

Stiffness matrix,  $\mathbf{K} : \begin{matrix} \mathbf{K}^{(1)}_{(8 \times 8)}, \mathbf{K}^{(2)}_{(8 \times 8)}, \mathbf{K}^{(3)}_{(8 \times 8)}, \mathbf{K}^{(4)}_{(8 \times 8)} \end{matrix} \rightarrow \mathbf{K}_{(18 \times 18)}$

$$\begin{matrix} u_1^{(2)} & u_2^{(2)} \\ \delta u_1^{(2)} & \left[ \begin{matrix} K_{11}^{(2)} & K_{12}^{(2)} & \dots \\ K_{21}^{(2)} & K_{22}^{(2)} & \dots \\ \vdots & \vdots & \ddots \end{matrix} \right] \\ \mathbf{K}_{8 \times 8}^{(2)} = \delta u_2^{(2)} & \end{matrix}$$

$$\begin{matrix} U_1 & U_2 & \cdots & U_5 & \cdots & U_{11} & \cdots & U_{18} \\ \delta U_1 & & & & | & & | & \\ \delta U_2 & & & & | & & | & \\ \vdots & & & & | & & | & \\ \mathbf{K}_{18 \times 18} = \delta U_{11} & - & - & - & K_{12}^{(2)} & - & K_{11}^{(2)} & - & - & - \\ \vdots & & & & | & & | & \\ \delta U_{18} & & & & | & & | & \end{matrix}$$

Load vector:  $\mathbf{R}$



$$\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S$$

$$\mathbf{R}_B = \sum_m \mathbf{R}_B^{(m)} \quad \text{with} \quad \mathbf{R}_B^{(m)} = \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{(m)} dV^{(m)}$$

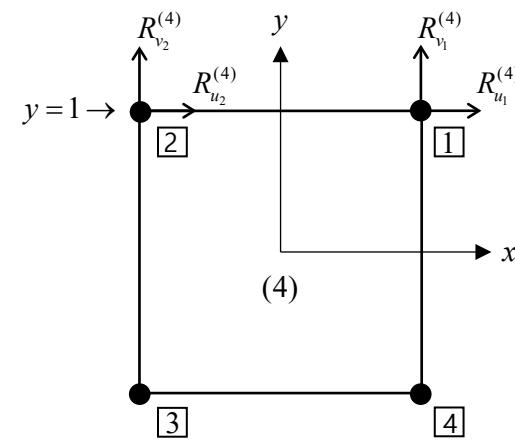
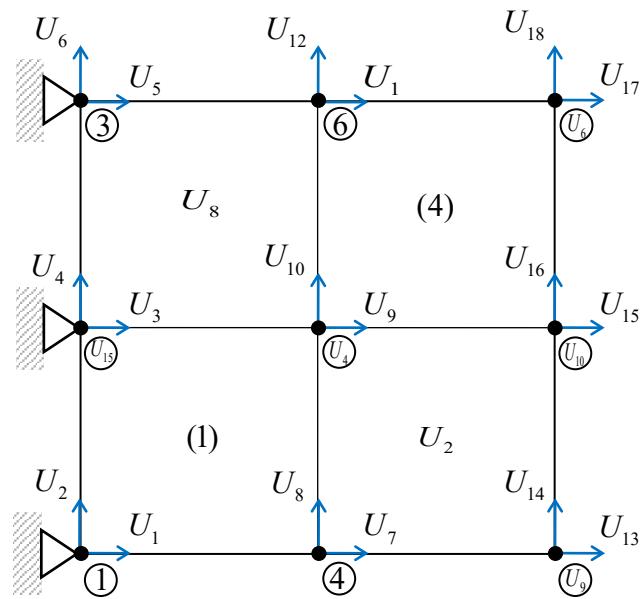
$$\mathbf{R}_S = \sum_m \mathbf{R}_S^{(m)} \quad \text{with} \quad \mathbf{R}_S^{(m)} = \int_{S_f^{(m)}} \mathbf{H}_S^{(m)T} \mathbf{f}^{S(m)} dS^{(m)}$$

$$\begin{bmatrix} u^{(4)} \\ v^{(4)} \end{bmatrix} = \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \mathbf{u}^{(4)} = \mathbf{H}^{(4)} \mathbf{u}^{(4)}$$

$$\begin{aligned} \mathbf{H}_s^{(4)} &= \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \text{ at } y=1 \\ &= \begin{bmatrix} (1+x)/2 & (1-x)/2 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & (1+x)/2 & (1-x)/2 & 0 & 0 \end{bmatrix} \end{aligned}$$

$$\mathbf{f}^{s(4)} = \begin{bmatrix} 0 \\ -0.5 \end{bmatrix}$$

$$\mathbf{R}^{(4)} = \mathbf{R}_S^{(4)} = \int_{S_f^{(4)}} \mathbf{H}_S^{(4)T} \mathbf{f}^{S(4)} dS^{(4)} = 1 \times \int_{-1}^1 \mathbf{H}_S^{(4)T} \mathbf{f}^{S(4)} dx = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ -0.5 \\ -0.5 \\ 0 \\ 0 \end{bmatrix}$$



$$\mathbf{R} = \begin{bmatrix} 0 \\ \vdots \\ 0 \\ -0.5 \\ 0.7 \\ \vdots \\ 0 \\ -0.5 \end{bmatrix} \leftarrow \begin{array}{l} R_{12} \\ R_{13} \\ \vdots \\ R_{18} \end{array}$$

We obtain the equilibrium equation.

$$\mathbf{KU} = \mathbf{R}$$

Imposition of displacement BC

$$\text{Displacement BC : } U_1 = U_2 = U_3 = U_4 = U_5 = U_6 = 0$$

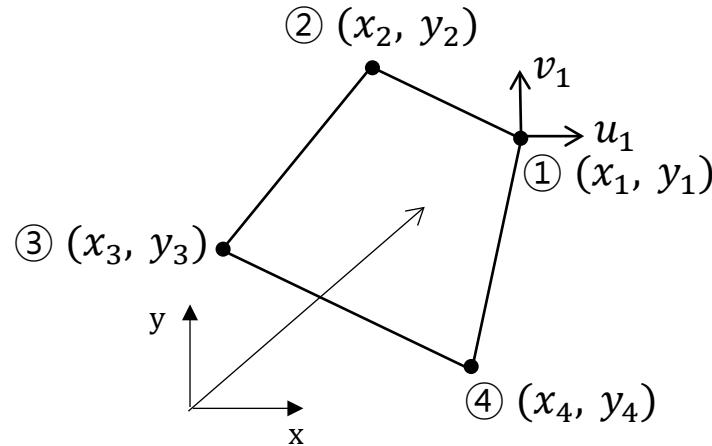
$$\tilde{\mathbf{K}}_{12 \times 12} \tilde{\mathbf{U}} = \tilde{\mathbf{R}}$$

Strain and stress

$$\begin{cases} \boldsymbol{\epsilon}^{(m)} = \mathbf{B}^{(m)} \mathbf{u}^{(m)} \\ \boldsymbol{\tau}^{(m)} = \mathbf{C}^{(m)} \mathbf{B}^{(m)} \mathbf{u}^{(m)} \end{cases}$$

## 6. Isoparametric Finite Element Procedure

Let us consider a 2D 4-node element  $m$



Interpolation of geometry

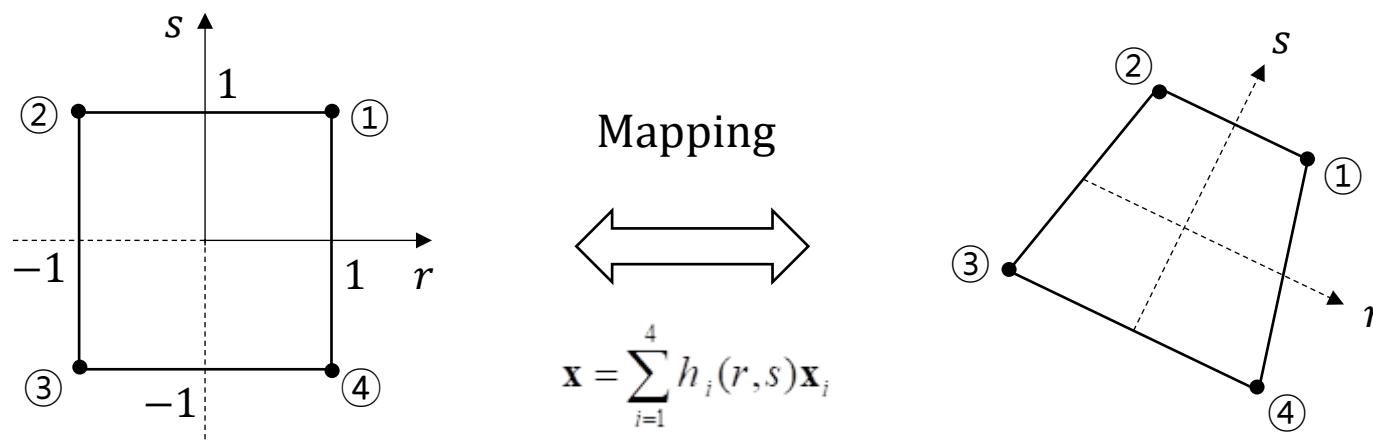
$$x^{(m)} = \sum_{i=1}^4 h_i x_i, \quad y^{(m)} = \sum_{i=1}^4 h_i y_i, \quad \text{where } x_i \text{ and } y_i \text{ are nodal coordinates.}$$

Interpolation of displacements

$$u^{(m)} = \sum_{i=1}^4 h_i u_i, \quad v^{(m)} = \sum_{i=1}^4 h_i v_i, \quad \text{where } u_i \text{ and } v_i \text{ are nodal displacements.}$$

The same interpolation functions ( $h_i$ ) are used for geometry and displacements.

Shape functions are defined in the natural coordinate system:  $h_i = h_i(r, s)$



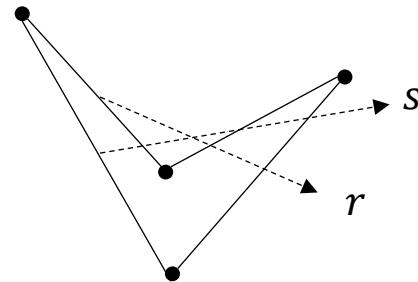
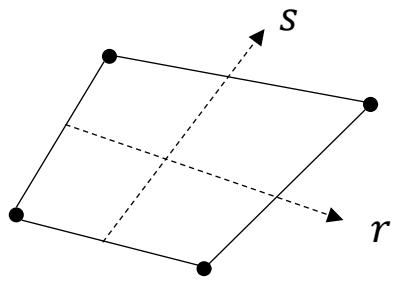
< Natural coordinate system >

$$h_1 = \frac{1}{4}(1+r)(1+s), \quad h_2 = \frac{1}{4}(1-r)(1+s), \quad h_3 = \frac{1}{4}(1-r)(1-s), \quad h_4 = \frac{1}{4}(1+r)(1-s)$$

< Global coordinate system >

Since the same shape functions are used for coordinates and displacements, we call the element an isoparametric element.

(Note) The element must give a unique correspondence between  $(r,s)$  and  $(x,y)$ .



$$\mathbf{u}(r, s) = \begin{bmatrix} u^{(m)} \\ v^{(m)} \end{bmatrix} = \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ v_1 \\ v_2 \\ v_3 \\ v_4 \end{bmatrix} = \mathbf{H}^{(m)} \mathbf{u}^{(m)}$$

In order to construct  $\mathbf{B}^{(m)}$  and  $\mathbf{K}^{(m)}$ , we need to calculate  $\frac{\partial u^{(m)}}{\partial x^{(m)}}$ ,  $\frac{\partial u^{(m)}}{\partial y^{(m)}}$ ,  $\frac{\partial v^{(m)}}{\partial x^{(m)}}$  and  $\frac{\partial v^{(m)}}{\partial y^{(m)}}$ .

(Note) If  $u^{(m)}$  is a function of  $x^{(m)}$ , we can directly calculate  $\frac{\partial u^{(m)}}{\partial x^{(m)}}$ . However,  $u^{(m)}$  is not a function of  $x^{(m)}$ , but a function of  $r$  and  $s$ .

## Jacobian matrix $\mathbf{J}$

$$u^{(m)} = u^{(m)}(r, s) \text{ and } v^{(m)} = v^{(m)}(r, s)$$

$$x^{(m)} = x^{(m)}(r, s) \text{ and } y^{(m)} = y^{(m)}(r, s)$$

"Chain rule"

$$\frac{\partial u^{(m)}}{\partial r} = \frac{\partial x^{(m)}}{\partial r} \frac{\partial u^{(m)}}{\partial x^{(m)}} + \frac{\partial y^{(m)}}{\partial r} \frac{\partial u^{(m)}}{\partial y^{(m)}}$$

$$\frac{\partial u^{(m)}}{\partial s} = \frac{\partial x^{(m)}}{\partial s} \frac{\partial u^{(m)}}{\partial x^{(m)}} + \frac{\partial y^{(m)}}{\partial s} \frac{\partial u^{(m)}}{\partial y^{(m)}}$$

$$\begin{bmatrix} \frac{\partial u^{(m)}}{\partial r} \\ \frac{\partial u^{(m)}}{\partial s} \end{bmatrix} = \begin{bmatrix} \frac{\partial x^{(m)}}{\partial r} & \frac{\partial y^{(m)}}{\partial r} \\ \frac{\partial x^{(m)}}{\partial s} & \frac{\partial y^{(m)}}{\partial s} \end{bmatrix} \begin{bmatrix} \frac{\partial u^{(m)}}{\partial x} \\ \frac{\partial u^{(m)}}{\partial y} \end{bmatrix} = \mathbf{J} \begin{bmatrix} \frac{\partial u^{(m)}}{\partial x} \\ \frac{\partial u^{(m)}}{\partial y} \end{bmatrix} \quad \text{with} \quad \mathbf{J} = \begin{bmatrix} \frac{\partial x^{(m)}}{\partial r} & \frac{\partial y^{(m)}}{\partial r} \\ \frac{\partial x^{(m)}}{\partial s} & \frac{\partial y^{(m)}}{\partial s} \end{bmatrix}$$

Similarly,

$$\begin{bmatrix} \frac{\partial v^{(m)}}{\partial r} \\ \frac{\partial v^{(m)}}{\partial s} \end{bmatrix} = \mathbf{J} \begin{bmatrix} \frac{\partial v^{(m)}}{\partial x} \\ \frac{\partial v^{(m)}}{\partial y} \end{bmatrix}.$$

$$\begin{bmatrix} \frac{\partial u^{(m)}}{\partial x^{(m)}} \\ \frac{\partial u^{(m)}}{\partial y^{(m)}} \end{bmatrix} = \mathbf{J}^{-1} \begin{bmatrix} \frac{\partial u^{(m)}}{\partial r} \\ \frac{\partial u^{(m)}}{\partial s} \end{bmatrix} \quad \text{and} \quad \begin{bmatrix} \frac{\partial v^{(m)}}{\partial x} \\ \frac{\partial v^{(m)}}{\partial y} \end{bmatrix} = \mathbf{J}^{-1} \begin{bmatrix} \frac{\partial v^{(m)}}{\partial r} \\ \frac{\partial v^{(m)}}{\partial s} \end{bmatrix}$$

## Strain-displacement relation

$$\begin{bmatrix} \frac{\partial u^{(m)}}{\partial x^{(m)}} \\ \frac{\partial u^{(m)}}{\partial y^{(m)}} \end{bmatrix} = \mathbf{J}^{-1} \begin{bmatrix} \frac{\partial u^{(m)}}{\partial r} \\ \frac{\partial u^{(m)}}{\partial s} \end{bmatrix}$$

$$\begin{bmatrix} \frac{\partial u^{(m)}}{\partial r} \\ \frac{\partial u^{(m)}}{\partial s} \end{bmatrix} = \begin{bmatrix} \frac{\partial h_1}{\partial r} & \frac{\partial h_2}{\partial r} & \frac{\partial h_3}{\partial r} & \frac{\partial h_4}{\partial r} & 0 & 0 & 0 & 0 \\ \frac{\partial h_1}{\partial s} & \frac{\partial h_2}{\partial s} & \frac{\partial h_3}{\partial s} & \frac{\partial h_4}{\partial s} & 0 & 0 & 0 & 0 \end{bmatrix} \mathbf{u}^{(m)}$$

with  $\mathbf{u}^{(m)} = [u_1 \quad u_2 \quad u_3 \quad u_4 \quad v_1 \quad v_2 \quad v_3 \quad v_4]^T$

$$\begin{bmatrix} \frac{\partial u^{(m)}}{\partial x^{(m)}} \\ \frac{\partial u^{(m)}}{\partial y^{(m)}} \end{bmatrix} = \mathbf{J}^{-1} \begin{bmatrix} \frac{\partial h_1}{\partial r} & \frac{\partial h_2}{\partial r} & \frac{\partial h_3}{\partial r} & \frac{\partial h_4}{\partial r} & 0 & 0 & 0 & 0 \\ \frac{\partial h_1}{\partial s} & \frac{\partial h_2}{\partial s} & \frac{\partial h_3}{\partial s} & \frac{\partial h_4}{\partial s} & 0 & 0 & 0 & 0 \end{bmatrix} \mathbf{u}^{(m)} = \begin{bmatrix} a_1 & a_2 & a_3 & a_4 & 0 & 0 & 0 & 0 \\ b_1 & b_2 & b_3 & b_4 & 0 & 0 & 0 & 0 \end{bmatrix} \mathbf{u}^{(m)}$$

Similarly,

$$\begin{bmatrix} \frac{\partial v^{(m)}}{\partial x^{(m)}} \\ \frac{\partial v^{(m)}}{\partial y^{(m)}} \end{bmatrix} = \mathbf{J}^{-1} \begin{bmatrix} 0 & 0 & 0 & 0 & \frac{\partial h_1}{\partial r} & \frac{\partial h_2}{\partial r} & \frac{\partial h_3}{\partial r} & \frac{\partial h_4}{\partial r} \\ 0 & 0 & 0 & 0 & \frac{\partial h_1}{\partial s} & \frac{\partial h_2}{\partial s} & \frac{\partial h_3}{\partial s} & \frac{\partial h_4}{\partial s} \end{bmatrix} \mathbf{u}^{(m)} = \begin{bmatrix} 0 & 0 & 0 & 0 & a_1 & a_2 & a_3 & a_4 \\ 0 & 0 & 0 & 0 & b_1 & b_2 & b_3 & b_4 \end{bmatrix} \mathbf{u}^{(m)}$$

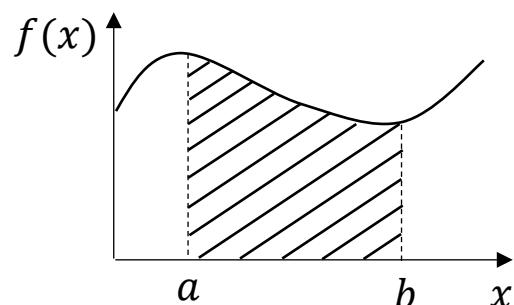
$$\begin{bmatrix} \varepsilon_{xx} \\ \varepsilon_{yy} \\ \gamma_{xy} \end{bmatrix} = \begin{bmatrix} \frac{\partial u^{(m)}}{\partial x^{(m)}} \\ \frac{\partial v^{(m)}}{\partial y^{(m)}} \\ \frac{\partial u^{(m)}}{\partial y^{(m)}} + \frac{\partial v^{(m)}}{\partial x^{(m)}} \end{bmatrix} = \begin{bmatrix} a_1 & a_2 & a_3 & a_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & b_1 & b_2 & b_3 & b_4 \\ b_1 & b_2 & b_3 & b_4 & a_1 & a_2 & a_3 & a_4 \end{bmatrix} \mathbf{u}^{(m)} = \mathbf{B}^{(m)}(r, s) \mathbf{u}^{(m)}$$

## Stiffness matrix

$$\mathbf{K}^{(m)} = \int_V \mathbf{B}^{(m)\top} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)} = t \int_{-1}^1 \int_{-1}^1 \mathbf{B}^{(m)\top}(r,s) \mathbf{C}^{(m)} \mathbf{B}^{(m)}(r,s) \det \mathbf{J} dr ds$$

with  $dV^{(m)} = \det \mathbf{J} \times t dr ds$

## Gauss Integration (Gaussian Quadrature)



$$\int_a^b f(x) dx = \sum_{i=1}^n w_i f(x_i)$$

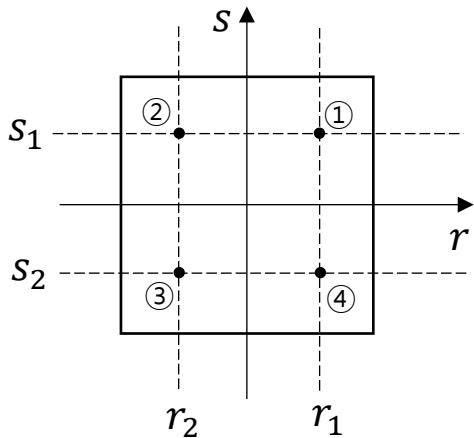
$x_i$ : Gauss points,  $w_i$ : Weight factors

2-point Gauss integration in 2D

$$\int_{-1}^1 f(x) dx = w_1 f(x_1) + w_2 f(x_2)$$

$$x_1 = -\frac{1}{\sqrt{3}}, \quad x_2 = \frac{1}{\sqrt{3}}, \quad w_1 = w_2 = 1$$

$$\int_{-1}^1 x^2 dx = 1 \times \left(-\frac{1}{\sqrt{3}}\right)^2 + 1 \times \left(\frac{1}{\sqrt{3}}\right)^2 = \frac{2}{3}$$



$$r_1 = s_1 = 1/\sqrt{3}$$

$$r_2 = s_2 = -1/\sqrt{3}$$

$$w_{ij} = 1$$

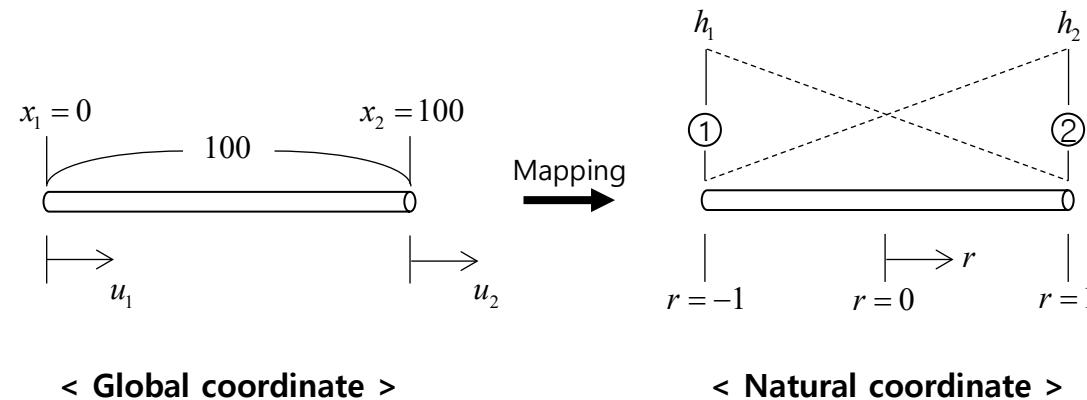
Using "2x2 Gauss integration" in 2D,

$$\mathbf{K}^{(m)} = t \sum_{i=1}^2 \sum_{j=1}^2 w_{ij} \mathbf{B}^{(m)\top}(r_i, s_j) \mathbf{C}^{(m)} \mathbf{B}^{(m)}(r_i, s_j) \det \mathbf{J}(r_i, s_j)$$

$$= w_{11} \mathbf{K}'(r_1, s_1) + w_{12} \mathbf{K}'(r_1, s_2) + w_{21} \mathbf{K}'(r_2, s_1) + w_{22} \mathbf{K}'(r_2, s_2)$$

$$\text{with } \mathbf{K}'(r, s) = t \mathbf{B}^{(m)\top}(r, s) \mathbf{C}^{(m)} \mathbf{B}^{(m)}(r, s) \det \mathbf{J}(r, s)$$

## Example – Stiffness of a 2-node bar finite element



$$u^{(m)} = \left(1 - \frac{x}{100}\right)u_1 + \frac{x}{100}u_2$$

$$\varepsilon_{xx} = \frac{du^{(m)}}{dx} = -\frac{1}{100}u_1 + \frac{1}{100}u_2 = \begin{bmatrix} -\frac{1}{100} & \frac{1}{100} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

$$\mathbf{K}^{(m)} = \int_0^{100} \mathbf{B}^{(m)T} E \mathbf{B}^{(m)} dx = \frac{E}{100} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

## Isoparametric procedure

$$x^{(m)} = \sum_{i=1}^2 h_i x_i = h_1 x_1 + h_2 x_2, \quad x^{(m)} = x^{(m)}(r)$$

$$u^{(m)} = \sum_{i=1}^2 h_i u_i = h_1 u_1 + h_2 u_2, \quad u^{(m)} = u^{(m)}(r) \quad \text{with } h_1 = \frac{1}{2}(1-r), \quad h_2 = \frac{1}{2}(1+r)$$

$$\varepsilon_{xx} = \frac{du^{(m)}}{dx^{(m)}}$$

$$\frac{dx^{(m)}}{dr} = \frac{d}{dr}(h_1 x_1 + h_2 x_2) = \frac{d}{dr}(h_1 \times 0 + h_2 \times 100) = \frac{d}{dr}\left(\frac{1}{2}(1+r) \times 100\right) = 50$$

$$\det \mathbf{J} = 50$$

$$\frac{du^{(m)}}{dr} = \frac{dx^{(m)}}{dr} \frac{du}{dx^{(m)}} = 50 \frac{du^{(m)}}{dx^{(m)}}$$

$$\frac{du^{(m)}}{dx^{(m)}} = \frac{1}{50} \frac{du^{(m)}}{dr}$$

$$\frac{du^{(m)}}{dr} = \begin{bmatrix} \frac{dh_1}{dr} & \frac{dh_2}{dr} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} -\frac{1}{2} & \frac{1}{2} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

$$\varepsilon_{xx} = \frac{du^{(m)}}{dx^{(m)}} = \frac{1}{50} \begin{bmatrix} -\frac{1}{2} & \frac{1}{2} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \mathbf{B}^{(m)} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

$$\mathbf{B}^{(m)}(r) = \frac{1}{50} \begin{bmatrix} -\frac{1}{2} & \frac{1}{2} \end{bmatrix}$$

$$\begin{aligned}\mathbf{K}^{(m)} &= \int_{-1}^1 \mathbf{B}^{(m)T}(r) E \mathbf{B}^{(m)}(r) \det \mathbf{J} dr \\ &= w_1 (\mathbf{B}^{(m)T} E \mathbf{B}^{(m)}) \Big|_{r=-\frac{1}{\sqrt{3}}} \times 50 + w_2 (\mathbf{B}^{(m)T} E \mathbf{B}^{(m)}) \Big|_{r=\frac{1}{\sqrt{3}}} \times 50\end{aligned}$$

with  $w_1 = w_2 = 1$

$$\mathbf{K}^{(m)} = 2(\mathbf{B}^{(m)T} E \mathbf{B}^{(m)}) \det \mathbf{J} = 2 \times \frac{1}{50} \begin{bmatrix} -\frac{1}{2} \\ \frac{1}{2} \end{bmatrix} E \begin{bmatrix} -\frac{1}{2} & \frac{1}{2} \end{bmatrix} \frac{1}{50} \times 50 = \frac{E}{100} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$